

AD 678569

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GAMD-731Q

Category A

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## TRIOIL

A THREE-DIMENSIONAL VERSION OF THE OIL CODE

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Advanced Research Projects Agency  
ARPA Order No. 71-62  
Ballistic Research Laboratories  
Contract No. DA-04-495-AMC-1481(X)  
GA Project 6003

June 1, 1967

## FOREWORD

The TRIOIL computer code described herein is as it existed on July 1, 1967. The code has been in continuous development for three years and in its presented form has been applied in this report. However, the development and improvement in both the physics and mathematics of the code are being continued, so that duplication of results (or even close agreement) between problems run with the code as published and the code as it existed either before or after this time is not necessarily to be expected.

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## 1. INTRODUCTION

The TRIOIL code is a three-dimensional continuous Eulerian cartesian hydrodynamic code.\* It is a natural extension of the two-dimensional OIL code (Ref. 1). The code is in an infant stage as of now but work is continuing to make it an operative tool for research in the field of numerical hydrodynamics. In this version of TRIOIL, the scalar pressure is the only stress. However, it appears to be straightforward to incorporate stresses due to strength and viscous forces as has been done for the two-dimensional version (Ref. 2) and adding two materials (Ref. 4).

The maximum size of the three-dimensional grid is limited by the present 32K to 64K storage computers. In this present version, the maximum number of cells in any single direction \*IMAX for the x-direction, JMAX for the y-direction and KMAX for the z-direction) is limited to 30; further, the total number of cells is limited to 6000. But, with high speed disks and drums, as are currently being made available, one can store x-y slabs of data. This means, at any given time, we have in core memory only two slabs of x-y data, which is sufficient for the various phases of the program. This procedure will essentially remove the limitations on grid size, and enable one to apply this code to solving many types of three-dimensional problems.

A rezone routine, that increases all linear cell dimensions by a factor of two, is presently an operative feature of this code. This routine is programmed primarily for hypervelocity impact calculations into an infinite target. To use it for different applications will require some minor modifications. Eight cells in the old grid are combined into one for the new grid. Energy, mass, and momentum are conserved. This routine assumes the following:

- (a) That IMAX, JMAX and KMAX are even integers. (iMAX, jMAX and kMAX are the total number of cells in the x, y, and z directions.)

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\* This work was funded jointly by the Systems and Research Department of Honeywell, Inc., at Minneapolis, Minnesota and General Atomic Division of General Dynamics Corporation.

- (b) That the  $j$  value of the initial interface between projectile and target is  $2^N$ , allowing  $N$  rezones.

This routine can be called for whenever mass leaves any of the six grid boundaries.

The six slides of the grid can either be reflective or transmittive. Variable zoning of  $\Delta x$ ,  $\Delta y$  and  $\Delta z$  is a working feature of the code. Active grid counters are calculated for all three directions. By doing this, one processes only that portion of the grid that is active.

Programs for displaying density, velocities and pressures are currently being programmed using the Stromberg-Carlson 4020 Plotter.

A subroutine called SETUP is available for generating the starting data for the TRIOIL code. This routine will only generate a rectangular parallelepiped for the projectile and target. For the more complicated geometries, we plan to modify the CLAM code (Ref. 1) to generate three-dimensional starting conditions for the TRIOIL code.

An accurate estimate of computer time required to run a realistic problem in three dimensions is not presently available, however, calculations reported here, have been run on the UNIVAC-1108 and Control Data 6600 computers to a point where the peak pressures have been attenuated by several orders of magnitude for computer times of approximately one hour.

The code is written in FORTRAN IV language and is operative on the IBM-7044 and the UNIVAC-1108 as well as the Control Data 6600.

## 2.1 Basic Equations

The Eulerian equations in cartesian geometry are:

$$(A) \frac{\partial \rho}{\partial t} + \frac{\partial \rho u}{\partial x} + \frac{\partial \rho v}{\partial y} + \frac{\partial \rho w}{\partial z} = 0$$

$$(B) \rho \frac{\partial u}{\partial t} + \rho u \frac{\partial u}{\partial x} + \rho v \frac{\partial u}{\partial y} + \rho w \frac{\partial u}{\partial z} + \frac{\partial P}{\partial x} = 0$$

$$\rho \frac{\partial v}{\partial t} + \rho u \frac{\partial v}{\partial x} + \rho v \frac{\partial v}{\partial y} + \rho w \frac{\partial v}{\partial z} + \frac{\partial P}{\partial y} = 0$$

$$\rho \frac{\partial w}{\partial t} + \rho u \frac{\partial w}{\partial x} + \rho v \frac{\partial w}{\partial y} + \rho w \frac{\partial w}{\partial z} + \frac{\partial P}{\partial z} = 0 .$$

$$(C) \quad \rho \frac{\partial E}{\partial t} + \rho u \frac{\partial E}{\partial x} + \rho v \frac{\partial E}{\partial y} + \rho w \frac{\partial E}{\partial z} + \frac{\partial (P_u)}{\partial x} + \frac{\partial (P_v)}{\partial y} + \frac{\partial (P_w)}{\partial z} = 0 .$$

These equations are solved in two parts, as in the OIL code (Ref. 1) and the familiar particle-in-cell codes (Ref. 3). The transport terms on the left side of Eq. (B) and (C) are temporarily dropped, while we compute (in PHI) the momentum and energy contributions due to pressure forces only. The omitted transport contributions to the momentum and energy are later approximated when we solve equation (A) and move mass across the cell boundaries.

#### 2.1.1. Effect of Pressure Forces Only (PHI)

Re-writing Eqs. (B) and (C) with the transport terms dropped

$$\rho \frac{\partial u}{\partial t} = - \frac{\partial P}{\partial x} \quad (1)$$

$$\rho \frac{\partial v}{\partial t} = - \frac{\partial P}{\partial y} \quad (2)$$

$$\rho \frac{\partial w}{\partial t} = - \frac{\partial P}{\partial z} \quad (3)$$

$$\rho \frac{\partial E}{\partial t} = - \frac{\partial P_u}{\partial x} - \frac{\partial P_v}{\partial y} - \frac{\partial P_w}{\partial z} \quad (4)$$

$$p = f(\rho, l) \quad \text{equation of state} \quad (5)$$

These variables, in one consistent set of units are

$\rho$  = density of cell (L) in g/cm<sup>3</sup>.

$x$  = x coordinate in cm.

$y$  = y coordinate in cm.

$z$  = z coordinate in cm.

$u$  = x component of velocity in cm/shake.

$v$  = y component of velocity in cm/shake.

$w$  = z component of velocity in cm/shake.

$P$  = material pressure in jerks/cm<sup>3</sup>.

$E$  = total specific energy in jerks/g.

$I$  = specific internal energy in jerks/g

$t$  = time in shakes

1 shake =  $10^{-8}$  sec

1 jerk =  $10^{16}$  ergs

There are no built-in units for this code, the user can select his own units in a given application by way of the equation of state constants and input data.

The density, pressure, velocities and internal energy are all cell-centered quantities referred to with the index (L).

Rewriting Eq. 4:

$$\rho \frac{\partial E}{\partial t} = - \frac{\partial P_u}{\partial x} - \frac{\partial P_v}{\partial y} - \frac{\partial P_w}{\partial z}$$

or

$$\rho \frac{\partial}{\partial t} [I + \frac{1}{2}(u^2 + v^2 + w^2)] = - \frac{\partial P_u}{\partial x} - \frac{\partial P_v}{\partial y} - \frac{\partial P_w}{\partial z}$$

$$\rho \frac{\partial I}{\partial t} + \rho u \frac{\partial u}{\partial t} + \rho v \frac{\partial v}{\partial t} + \rho w \frac{\partial w}{\partial t} = - P \frac{\partial u}{\partial x} - u \frac{\partial P}{\partial x} - P \frac{\partial v}{\partial y} - v \frac{\partial P}{\partial y} - P \frac{\partial w}{\partial z} - w \frac{\partial P}{\partial z}$$

but

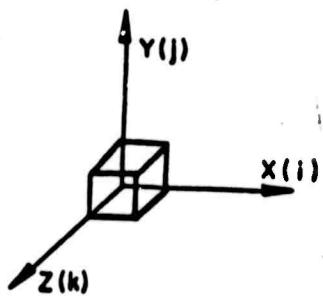
$$\rho \frac{\partial u}{\partial t} = - \frac{\partial P}{\partial x} \quad \text{and} \quad \rho \frac{\partial v}{\partial t} = - \frac{\partial P}{\partial y} \quad \text{and} \quad \rho \frac{\partial w}{\partial t} = - \frac{\partial P}{\partial z}$$

thus

$$\rho \frac{\partial I}{\partial t} = - P \left[ \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right] \quad (6)$$

Now we can write the four differential equations (1, 2, 3 and 6) in difference form.

The storage arrays for the cell centered quantities (mass, velocities, pressure and specific internal energy are as follows (Fig. 1).

 $J_{MAX}$ 

13	14	15	16
9	10	11	12
5	6	7	8
1	2	3	4

 $I_{MAX}$ 

29	30	31	32
25	26	27	28
21	22	23	24
17	18	19	20

 $K = 2$ 

45	46	47	48
41	42	43	44
37	38	39	40
33	34	35	36

 $K = 3$ 

61	62	63	64
57	58	59	60
53	54	55	56
49	50	51	52

 $K = 4$  $(K_{MAX})$ 

(L) THE INDEX OF THE CELL  
IS DEFINED AS =

$$L = (j-1) I_{MAX} + i + (k-1) I_{X MAX}$$

$$I_{X MAX} = (I_{MAX}) (J_{MAX})$$

Fig. 1

In the discussion to follow, please refer to Fig. 2. Re-writing the equations (1 to 3) in difference form results in:

The x-momentum equation (1) becomes in difference form

$$\rho_L^n \left( \frac{\tilde{u}_L^n - u_L^n}{\Delta t} \right) = \frac{P_{L-1}^n - P_{L+1}^n}{2\Delta x(i)}$$

The y-momentum equation (2) becomes in difference form

$$\rho_L^n \left( \frac{\tilde{v}_L^n - v_L^n}{\Delta t} \right) = \frac{P_{L-1\text{MAX}}^n - P_{L+1\text{MAX}}^n}{2\Delta y(j)}$$

The z-momentum equation (3) becomes in difference form

$$\rho_L^n \left( \frac{\tilde{w}_L^n - w_L^n}{\Delta t} \right) = \frac{P_{L-(i\text{MAX})(j\text{MAX})}^n - P_{L+(i\text{MAX})(j\text{MAX})}^n}{2\Delta z(k)}$$

Here the acceleration of cell L is seen to depend only on the pressures in the neighbor cells (not that of L). Defining pressures at interfaces,

$$P_L^n = \frac{P_{L-1}^n + P_L^n}{2}$$

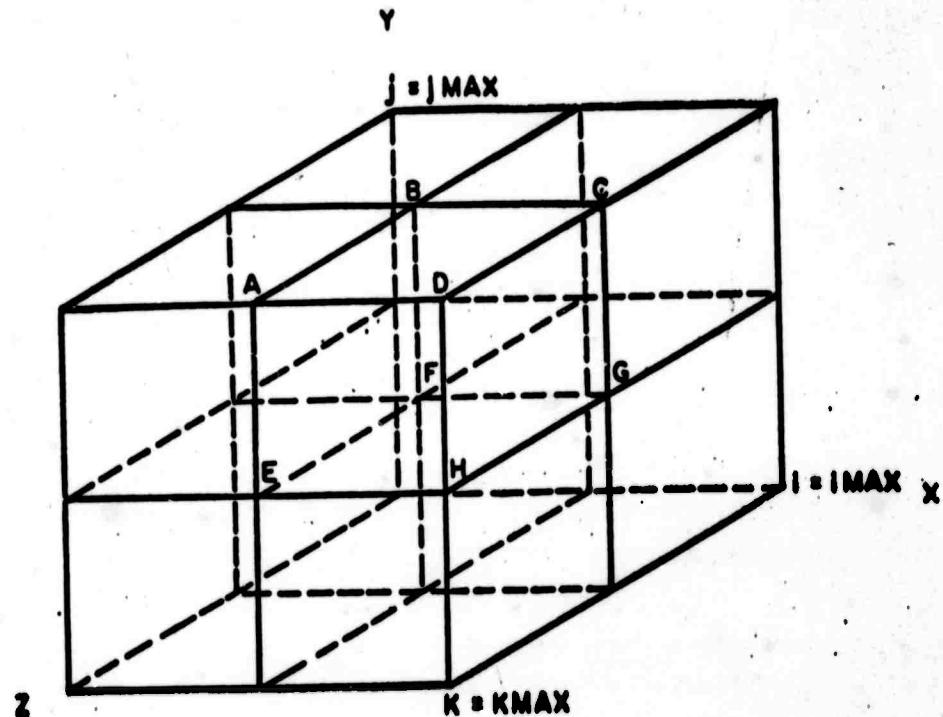
$$PRR^n = \frac{P_L^n + P_{L+1}^n}{2}$$

$$PBL0^n = \frac{P_{L-1\text{MAX}}^n + P_L^n}{2}$$

$$PABOVE^n = \frac{P_L^n + P_{L+1\text{MAX}}^n}{2}$$

$$PBIND^n = \frac{P_{L-(i\text{MAX})(j\text{MAX})}^n + P_L^n}{2}$$

$$PZR^n = \frac{P_{L+(i\text{MAX})(j\text{MAX})}^n + P_L^n}{2}$$



(L) THE INDEX OF THE CELL  
IS DEFINED AS =

$$L = (j-i)IMAX + i + (k-i)IXMAX$$

$$IXMAX = (iMAX)(jMAX)$$

$$X(i) = \sum_{j=1}^i \Delta X(j)$$

$$Y(j) = \sum_{i=1}^j \Delta Y(i)$$

$$Z(k) = \sum_{K=1}^k \Delta Z(K)$$

Fig. 2

Substituting these interface pressures into the three-momentum equations results in:

$$\tilde{u}_L - u_L^n = \frac{\Delta t}{\rho_L^n} \left[ \frac{P_L^n - P_{RR}^n}{\Delta x(i)} \right]$$

$$\tilde{v}_L - v_L^n = \frac{\Delta t}{\rho_L^n} \left[ \frac{P_{BLO}^n - P_{ABOVE}^n}{\Delta y(j)} \right]$$

$$\tilde{w}_L - w_L^n = \frac{\Delta t}{\rho_L^n} \left[ \frac{P_{BIND}^n - P_{ZR}^n}{\Delta z(k)} \right]$$

where the usual  $\sim$  (tilda) designates the new velocities (not at cycle  $n+1$ ), since we have temporarily dropped the transport terms).

The specific internal energy equation becomes

$$\rho_L \frac{\tilde{I}_L - I_L^n}{\Delta t} = - P_L^n \left[ \frac{v_{L-iMAX}^{n+\frac{1}{2}} - v_{L+iMAX}^{n+\frac{1}{2}}}{2\Delta y(j)} + \frac{u_{L-1}^{n+\frac{1}{2}} - u_{L+1}^{n+\frac{1}{2}}}{2\Delta x(i)} \right. \\ \left. + \frac{w_{L-(iMAX)(jMAX)}^{n+\frac{1}{2}} - w_{L+(iMAX)(jMAX)}^{n+\frac{1}{2}}}{2\Delta z(k)} \right]$$

The reason for the velocities at time  $(n+\frac{1}{2})$  can be seen from energy conservation considerations, which we will discuss later in the text. Defining

$$u_L^{n+\frac{1}{2}} = \frac{u_L^n + \tilde{u}_L}{2}$$

$$v_L^{n+\frac{1}{2}} = \frac{v_L^n + \tilde{v}_L}{2}$$

$$w_L^{n+\frac{1}{2}} = \frac{w_L^n + \tilde{w}_L}{2}$$

and

$$VBLO = \frac{v_L + v_{L-1\text{MAX}}}{2}.$$

$$VABOVE = \frac{v_L + v_{L+1\text{MAX}}}{2}.$$

$$UL = \frac{U_L + U_{L-1}}{2}.$$

$$URR = \frac{U_L + U_{L+1}}{2}.$$

$$UBIND = \frac{W_L + W_{L-(i\text{MAX})(j\text{MAX})}}{2}.$$

$$WZR = \frac{W_L + W_{L+(i\text{MAX})(j\text{MAX})}}{2}.$$

then

$$\begin{aligned} \rho_L^n \left( \frac{\tilde{I}_L - I_L^n}{\Delta t} \right) = P_L^n & \left[ \frac{VBLO^n + VBL0}{2\Delta y(j)} - \frac{VABOVE^n + VABOVE}{2\Delta y(j)} + \frac{UL^n + \tilde{U}_L}{2\Delta X(i)} \right. \\ & \left. - \frac{URR^n + \tilde{U}_R}{2\Delta X(i)} + \frac{UBIND^n + UBIND}{2\Delta Z(k)} - \frac{WZR^n + \tilde{W}_R}{2\Delta Z(k)} \right] \end{aligned}$$

The solution of the momentum equations are very straightforward, however, the solution to the energy equation requires the velocities at two different time steps (the old and new velocities). We have chosen to make two passes through the routine, the first pass to integrate the momentum equations, but formulate the interface velocities first for the work term contribution to the internal energy before integrating the momentum equations (since we have allowed only one array per velocity component).

The second pass, we bypass the momentum equations, and only compute the new interface velocities for their contribution to the work term. A single pass can be done by looking ahead two cells above; two cells to the right and two cells in front.

The choice of the velocities at  $n + \frac{1}{2}$  in the energy equation is apparent in the following discussion. For convenience we will go through the logic in the  $y$  direction. Since we have dropped the transport terms, our integration of the momentum and energy equations have not been advanced to time ( $n + 1$ ). As before, we designate the phase 1 velocities and energy as  $\tilde{v}$ ,  $\tilde{v}$ ,  $w$ , and  $\tilde{I}$ .

$$\tilde{v}_{j-\frac{1}{2}} = v_{j-\frac{1}{2}}^n + \frac{\Delta t}{\rho_{j-\frac{1}{2}}^n} \left[ \frac{P_{j-3/2}^n - P_{j+\frac{1}{2}}^n}{2\Delta y(j)} \right]$$

and

$$\tilde{I}_{j-\frac{1}{2}} = I_{j-\frac{1}{2}}^n + \frac{\Delta t}{\rho_{j-\frac{1}{2}}^n} \left[ \frac{\bar{v}_{j-3/2} - \bar{v}_{j+\frac{1}{2}}}{2\Delta y(j)} \right]$$

where

$$\bar{v}_{j-3/2} = \frac{\tilde{v}_{j-3/2} + v_{j-3/2}^n}{2}$$

and

$$\bar{v}_{j+\frac{1}{2}} = \frac{\tilde{v}_{j+\frac{1}{2}} + v_{j+\frac{1}{2}}^n}{2}$$

Before entering PH1, where the quantities are at time  $n$ , the total energy of the system (considering one dimension, the  $y$  direction) is:

$$E^n = \sum_{j=1}^{JMAX} AMX_{j-\frac{1}{2}}^n \left[ I_{j-\frac{1}{2}}^n + \frac{1}{2} \left( v_{j-\frac{1}{2}}^n \right)^2 \right]$$

And the total energy at the end of PH1 is then

$$\tilde{E} = \sum_{j=1}^{JMAX} AMX_{j-\frac{1}{2}}^n \left[ \tilde{I}_{j-\frac{1}{2}} + \frac{1}{2} \left( \tilde{v}_{j-\frac{1}{2}} \right)^2 \right]$$

The change,  $\Delta E = E^n - \tilde{E}$  should be zero for energy conservation or

$$\Delta E = \sum_{j=1}^{JMAX} AMX_{j-\frac{1}{2}}^n \left[ I_{j-\frac{1}{2}}^n - \tilde{I}_{j-\frac{1}{2}} + \frac{1}{2} \left( v_{j-\frac{1}{2}}^n \right)^2 - \frac{1}{2} \left( \tilde{v}_{j-\frac{1}{2}} \right)^2 \right]$$

the  $\Delta$  kinetic terms can be represented by

$$\begin{aligned}
 & \frac{(v_{j-\frac{1}{2}}^n + \tilde{v}_{j-\frac{1}{2}})}{2} \left[ v_{j-\frac{1}{2}}^n - \tilde{v}_{j-\frac{1}{2}} \right] \\
 \Delta E = & \sum_{j=1}^{j_{\text{MAX}}} AMX_{j-\frac{1}{2}}^n \left[ I_{j-\frac{1}{2}}^n - \bar{I}_{j-\frac{1}{2}} + \bar{v}_{j-\frac{1}{2}} (v_{j-\frac{1}{2}}^n - \tilde{v}_{j-\frac{1}{2}}) \right] \\
 = & \sum_{j=1}^{j_{\text{MAX}}} AMX_{j-\frac{1}{2}}^n \left[ - \frac{\Delta t}{\rho_{j-\frac{1}{2}}^n} \frac{P_{j-\frac{1}{2}}^n}{2 \Delta y_j} \left( \frac{\bar{v}_{j-3/2} - \bar{v}_{j+\frac{1}{2}}}{2 \Delta y_j} \right) \right. \\
 & \left. - \bar{v}_{j-\frac{1}{2}} \frac{\Delta t}{\rho_{j-\frac{1}{2}}^n} \left( \frac{P_{j-3/2}^n - P_{j+\frac{1}{2}}^n}{2 \Delta y_j} \right) \right] \\
 = & \Delta t \sum_{j=1}^{j_{\text{MAX}}} \frac{AMX_{j-\frac{1}{2}}^n}{\rho_{j-\frac{1}{2}}^n 2 \Delta y_j} \left[ - P_{j-\frac{1}{2}}^n \bar{v}_{j-3/2} \right. \\
 & \left. + P_{j-\frac{1}{2}}^n \bar{v}_{j+\frac{1}{2}} - P_{j-3}^n \bar{v}_{j-\frac{1}{2}} + P_{j+\frac{1}{2}}^n \bar{v}_{j-\frac{1}{2}} \right] \\
 = & \frac{-\Delta t}{2} DX(i) DZ(k) \sum_{j=1}^{j_{\text{MAX}}} \left[ P_{j-\frac{1}{2}}^n \bar{v}_{j-3/2} + P_{j-3/2}^n \bar{v}_{j-\frac{1}{2}} \right. \\
 & \left. - P_{j+\frac{1}{2}}^n \bar{v}_{j-\frac{1}{2}} - P_{j-\frac{1}{2}}^n \bar{v}_{j+\frac{1}{2}} \right]
 \end{aligned}$$

Thus, the last two terms in  $j$  being cancelled by the first two terms in  $j+1$ . By prescribing the proper boundary conditions, we will have energy conservation for the entire grid.

**EXAMPLE:**for  $j = 1$ 

$$\begin{array}{c}
 P_{1/2} v_{-1/2} + P_{-1/2} v_1 - P_{3/2} v_{1/2} - P_{1/2} v_{3/2} \\
 \downarrow \quad \downarrow \quad \downarrow \quad \downarrow \\
 j = 2 \\
 P_{3/2} v_{1/2} + P_{1/2} v_{3/2} - P_{5/2} v_{3/2} - P_{3/2} v_{5/2} \\
 \downarrow \quad \downarrow \quad \downarrow \quad \downarrow \\
 j = 3 \\
 P_{5/2} v_{3/2} + P_{3/2} v_{5/2} - P_{7/2} v_{5/2} - P_{5/2} v_{7/2}
 \end{array}$$

Note, the boundary terms do not cancel. For the example, assume the bottom boundary to be reflective. Referring to Eq. (A), the first two terms do not cancel. If, however we set the pressure of the mirror cell  $P_{-1/2}^n = P_{1/2}^n$  (this does not imply that  $v_{-1/2} = 0$ ) and  $v_{-1/2} = -v_1$ , this does lead to cancellation of the first two terms. If, however, we designate the bottom boundary to be transmittive, our boundary conditions are  $v_{-1/2} = 0_j$  which implies  $P_{-1/2}^n = P_{3/2}^n$ , and that  $v_{-1/2} = v_1$ . This still leaves us with the two terms  $P_{1/2} v_{-1/2} + P_{3/2} v_1$ , thus we compensate for these terms, by adding or subtracting energy to the system. To keep all the books straight,  $E$  th must be modified accordingly. Similar boundary conditions may exist for the other four sides of the grid.

**2.1.2 Adding the Transport Terms (PH2)**

Rewriting Eq. (A) the mass transport equation in finite difference form results in

$$\frac{\rho_L^{n+1} - \rho_L^n}{\Delta t} = \left[ \frac{\rho_{j-1}^n \tilde{v}_{j-1} - \rho_j^n \tilde{v}_j}{\Delta y(j)} + \frac{\rho_{i-1}^n \tilde{u}_{i-1} - \rho_i^n \tilde{u}_i}{\Delta x(i)} + \frac{\rho_{k-1}^n \tilde{w}_{k-1} - \rho_k^n \tilde{w}_k}{\Delta z(k)} \right]$$

or in the formulation for mass we have

$$\Delta M_L = \Delta t \left[ (Av)_B^y \rho - (Av)_T^y \rho + (Au)_L^X \rho + (Au)_R^X \rho + (Aw)_{BH}^Z \rho - (Aw)_F^Z \rho \right]$$

where the superscripts X, Y, Z, refer to the particular coordinate, and B = bottom, T = top, L = left, R = right, BH = behind and F = front. The mass that is moved from cell to cell, has then momentum and energy associated with it, thus these are the approximations to the transport terms that were omitted in PH1 for the momentum and energy equations.

The  $\rho$  (density) is that of the donor cell, and the velocity is yet to be determined. Various techniques for the velocity weighting have been tried (Ref. 1). The velocity weighting scheme in this report is identical to that in the OIL report (Ref. 1).

### 2.1.3 Time Control

The time step ( $\Delta t$ ) is controlled by the Courant condition and the condition that the mass flux equation will not empty more mass from a cell than is there.

Take the y direction (a similar treatment is done for the other two directions)

$$\Delta M_y = \bar{\rho} v A \Delta t$$

Let

$$\bar{v} = v(L)$$

$$\bar{\rho} = \rho(L)$$

$$\Delta M_y = AMX(L)$$

then

$$AMX(L) = \rho(L) v(L) A_j \Delta t$$

$$= \frac{AMX(L)}{DX(i) Dy(j) DZ(k)} v(L) DX(i) DZ(k) \Delta t$$

$$= \frac{AMX(L) v(L) \Delta t}{Dy(j)}$$

or  $|v_{(L)}| \Delta t \leq Dy(j)$  such that the flux in the y direction will not empty the cell (L).

The Courant condition is that  $\frac{C}{C_M} \leq 1$  where  $C$  = sound speed and  $C_M$  is the maximum speed at which a disturbance can propagate in the given grid or  $C \frac{\Delta t}{\Delta x} \leq 1.$ , or  $C \frac{\Delta t}{\Delta y} \leq 1.$ , and  $C \frac{\Delta t}{\Delta z} \leq 1.$

#### 2.1.4 Remarks

No corner coupling (that is, mass is constrained to move at right angles to the sides of the cell) exists in this version, and no attempt to systematically study this has been initiated. The movement of mass across the cell boundaries give rise to forces which are effective in reducing fluctuations that arise from the differencing technique. That is of the form of a "true" viscosity, being proportional to the velocity gradient. It is this force which enables the Eulerian codes to treat shocks, where again, as in the Q method used in Lagrangian codes, the shock is spread over two or three zones.

### 2.2 Logic of TRIOIL

The logic involved in following a given cell (L) from  $t$  to  $t + \Delta t$  or cycle  $n$  to  $n + 1$  is as follows:

We assume we have integrated the mass, the three velocity components and the internal energy to cycle  $n$ , now all that remains to complete cycle  $n$ , and to begin cycle  $n + 1$ , is to update the pressures from the equation of state and calculate a new time step.

#### 2.2.1 CDT Routine

Here we calculate the pressure ( $P$ ) array for the entire grid. The pressure ( $P_L$ ) =  $f(\rho_L, I_L)$  where  $L$  is the index of the cell in question, defined as  $L = (j-1) iMAX + i + (k-1) (iMAX) (jMAX)$ . The density ( $\rho_L = \frac{AMX(L)}{DX(i) Dy(j) DZ(k)}$ ) is not one of the variables, it must be calculated several times during the cycle. It is planned in the future, to replace the mass storage with density. The speed of sound  $C = (\frac{\gamma P}{\rho})^{1/2}$  for a polytropic equation of state or  $(\frac{\partial P}{\partial \rho})_s^{1/2}$  for a general form is then calculated. From the particle velocities and speed of sound, a new  $\Delta t$  is calculated.

(Options for  $\Delta t$  control are identical to that in OIL.) The cycle number and the time are now advanced.

### 2.2.2 The EDIT Routine

This code has three separate editing routines all included in the subroutine EDIT. The first of these is the routine called "Short Print". This displays the problem number, cycle number, time, internal and kinetic energy, energy check and indices of the cell that is controlling the time step. The next routine available is called "Long Print". Here one edits each column, the three velocity components, pressure, mass, density, specific internal energy and  $y$ , all versus  $j$  (the index of the rows). Thus there are  $k_{MAX}$  sets of these, beginning with  $k=1$ . The normal units chosen are gram-cm-sh. which gives a logical unit for the pressure as  $10^{16}$  ergs/cm<sup>3</sup> and for the specific internal energy as  $10^{16}$  ergs/gram.

### 2.2.3 PH1 Routine

Here we integrate the three momentum equations and the internal energy equation due to pressure forces only. No material is moved at this time, and the transport terms are dropped temporarily. Using the new pressure and  $\Delta t$ , we can solve the momentum and energy equations.

$PL(j)$ , the pressure at interface ( $i-1$ ) and  $UL(j)$ , the velocity at interface ( $i-1$ ) are available from the previous column sweep on  $i-1$ :

$$PL(j) = \frac{P_L^n + P_{L-1}^n}{2}$$

$$UL(j) = \frac{U_L^n + U_{L-1}^n}{2}$$

The PBLO term which was the PABOVE for the cell below ( $L-i_{MAX}$ ) and VBLO which was VABOVE for cell below, are also available for interface  $j-1$ .

$$PBLO = \frac{P_L^n + P_{L-i_{MAX}}^n}{2}$$

$$VBLO = \frac{V_L^n + V_{L-i_{MAX}}^n}{2}$$

The PBIND term (the pressure at the back surface of the cell) which was PZR for cell L - (iMAX) (jMAX) and UBIND (the velocity at the back surface of the cell) which was WZR for cell L - (iMAX) (jMAX) are also available for interface K-1.

$$PBIND = \frac{P_L^n + P_{L-(iMAX)}^n}{2}$$

$$UBIND = \frac{W_L^n + W_{L-(iMAX)}^n}{2}$$

Thus, we need only to calculate quantities at the top, the right, and the front of cell (L). At the top we calculate

$$P ABOVE = \frac{P_L^n + P_{L+iMAX}^n}{2}$$

and V ABOVE =  $\frac{V_L^n + V_{L+iMAX}^n}{2}$ ; at the right we calculate PR =  $\frac{P_L^n + P_{L+1}^n}{2}$

and U RR as  $= \frac{U_L^n + U_{L+1}^n}{2}$  and in front as

$$PZR = \frac{P_L^n + P_{L+(iMAX)}^n}{2}$$

$$WZR = \frac{W_L^n + W_{L+(iMAX)}^n}{2}$$

When the cell in question is void, the pressures at the top, right and front interface are set to zero and the velocities are set = to the velocity of the cell above, to the right, and in front respectively. If a occupied cell has a void neighbor, the pressure at that interface is set = 0, and the velocity at that interface is set = to the velocity of the occupied cell in question.

We now have sufficient information to integrate the three momentum equations and part of the internal energy equation.

$$\rho \frac{\partial u}{\partial t} = -\frac{\partial p}{\partial x}$$

or

$$\bar{U}_L = U_L^n + \frac{P_L^n(j) - P_{RR}^n}{AMX_L^n} Dy(j) DZ(k) \Delta t$$

and

$$\rho \frac{\partial v}{\partial t} = -\frac{\partial p}{\partial y}$$

or

$$\tilde{v}_L^n = v_L^n + \frac{PBLO^n - PABOVE^n}{AMX_L^n} DX(i) DZ(k) \Delta t$$

and

$$\rho \frac{\partial w}{\partial t} = -\frac{\partial p}{\partial z}$$

or

$$\tilde{w}_L^n = w_L^n + \frac{(PBIND(M)^n - PZR^n)}{AMX_L^n} Dy(j) DX(i) \Delta t$$

where the index  $m$  refers to the  $(i, j)$  value of the slab  $x-y$  just behind. We can add the work terms due to velocities at cycle  $n$  to the change in internal energy as

$$\begin{aligned} \tilde{\gamma}_L^{(1)} = I_L^n &+ \frac{P_L^n \Delta t}{AMX_L^n} \left[ \frac{(VBLO^n - VABOVE^n)}{2} DX(i) DZ(k) \right. \\ &+ \left. \frac{(UL(j)^n - URR^n)}{2} Dy(j) DZ(k) \right. \\ &+ \left. \left. \frac{(UBIND(m)^n - WZR^n)}{2} DX(i) Dy(j) \right] \right. \end{aligned}$$

Now, one more pass is made through the entire grid, this time omitting the momentum equations but calculating the interface velocities, resulting in the integration of the internal energy to time ( $\sim$ ).

$$\begin{aligned} \tilde{\gamma}_L = \tilde{\gamma}_L^{(1)} &+ \frac{P_L^n \Delta t}{AMX_L^n} \left[ \frac{(VBLO - VABOVE)}{2} DX(i) DZ(k) \right. \\ &+ \left. \frac{(UL(j) - URR)}{2} Dy(j) DZ(k) \right. \\ &+ \left. \left. \frac{(UBIND(m) - WZR)}{2} DX(i) Dy(j) \right] \right. \end{aligned}$$

The option to integrate backwards from time ( $\sim$ ) to  $n$  if a negative internal energy is encountered, is not available in this version.

#### 2.2.4 PH2 Routine

Here, we move mass across the fixed boundaries. Momenta and energy is carried across with this mass and this approximates the transport terms omitted from the momentum and energy equations in PH1. Please refer to Fig. 2 for the following discussion.

The points A, B, C, D, E, F, G, and H are the eight corners of the cell. The following notation will be followed: side AEFB refers to the left, side ABCD refers to the top, CDHG refers to the right, BFCH to behind, EFGH to bottom, and ADEH to the front.

The five quantities associated with each interface are as follows:

<u>TOP</u>	AMPY      = mass crossing the top AMUT      = X momentum of this mass AMVT      = Y momentum of this mass AMWT      = Z momentum of this mass DELET      = specific energy across the top
<u>RIGHT</u>	AMMP      = mass crossing the right AMUR      = X momentum of this mass AMVR      = Y momentum of this mass AMWR      = Z momentum of this mass DELER      = specific energy across the right
<u>BOTTOM</u>	AMMY      = mass crossing the right AMMU      = X momentum of this mass AMMV      = Y momentum of this mass AMMW      = Z momentum of this mass DELEB      = specific energy across the bottom
<u>LEFT</u>	GAMC (j)      = mass crossing the left FLEFT (j)      = X momentum of this mass YAMC (j)      = Y momentum of this mass ZMOM (j)      = Z momentum of this mass SIGC (j)      = specific energy across the left
<u>BEHIND</u>	BMASS (M)      = mass crossing the back surface BXMOM (M)      = X momentum of this mass BYMOM (M)      = Y momentum of this mass BZMOM (M)      = Z momentum of this mass BENR (M)      = specific energy across the back surface

<u>FRONT</u>	FMASS      = mass crossing the front FXMOM     = X momentum of this mass FYMOM     = Y momentum of this mass FZMOM     = Z momentum of this mass FENR      = specific energy across the front
--------------	---

Following the typical cell ( $L$ ) the masses, the momenta, and the specific energies are now available at the left and bottom and back boundaries of cell ( $L$ ) from the previous column sweep, the cell below and the previous X-Y slab ( $K-1$ ).

The proper boundary conditions are first set for the cell ( $L=1$ ). We then begin by calculating

$$V ABOVE = \frac{\bar{V}_L + \bar{V}_{L+iMAX}}{2} \quad \text{for the top}$$

$$URR = \frac{\bar{U}_L + \bar{U}_{L+1}}{2} \quad \text{for the right}$$

$$W OUT = \frac{\bar{W}_L + \bar{W}_{L+(iMAX)(jMAX)}}{2} \quad \text{for in front}$$

Then form

$$V ABOVE = \frac{V ABOVE}{1 + (\bar{V}_{L+iMAX} - \bar{V}_L) \Delta t}$$

$$URR = \frac{URR}{1 + (\bar{U}_{L+1} - \bar{U}_L) \Delta t}$$

and

$$W OUT = \frac{W OUT}{1 + (\bar{W}_{L+(iMAX)(jMAX)} - \bar{W}_L) \Delta t}$$

Now we can calculate the mass crossing the three boundaries as

$$\Delta M_{TOP} = AMPY = \frac{AMX(M)^n V ABOVE \Delta t}{\Delta y(j)}$$

$$\Delta M_{RIGHT} = AMMP = \frac{AMX(M)^n URR \Delta t}{\Delta x(i)}$$

$$AM_{FRONT} = FMASS = \frac{AMX(M)^n WOUT \Delta t}{\Delta z(k)}$$

where (M) is the index of the donor cell. The donor cell is calculated from the sign of the weighted velocity.

The momenta associated with these three masses are now calculated where the velocity in the momenta is from the donor cell. The total specific energy that these mass fluxes carry are calculated at this time also. The momenta associated with the flux at the top is:

$$X \text{ component} = AMUT = AMPY [\bar{U}(N)]$$

$$Y \text{ component} = AMVT = AMPY [\bar{V}(N)]$$

$$Z \text{ component} = AMWT = AMPY [\bar{W}(N)]$$

$$\text{Specific energy} = DELET = \bar{T}_{(N)} + \frac{[\bar{U}_{(N)}^2 + \bar{V}_{(N)}^2 + \bar{W}_{(N)}^2]}{2}$$

Those associated with the flux at the right are:

$$X \text{ component} = AMUR = AMMP [\bar{U}(N)]$$

$$Y \text{ component} = AMVR = AMMP [\bar{V}(N)]$$

$$Z \text{ component} = AMWR = AMMP [\bar{W}(N)]$$

$$\text{Specific energy} = DELER = \bar{T}_{(N)} + \frac{[\bar{U}_{(N)}^2 + \bar{V}_{(N)}^2 + \bar{W}_{(N)}^2]}{2}$$

Those associated with the flux in front are:

$$X \text{ component} = FMASS [\bar{U}(N)] = FXMOM$$

$$Y \text{ component} = FMASS [\bar{V}(N)] = FYMOM$$

$$Z \text{ component} = FMASS [\bar{W}(N)] = FZMOM$$

$$\text{Specific energy} = \bar{T}_{(N)} + \frac{[\bar{U}_{(N)}^2 + \bar{V}_{(N)}^2 + \bar{W}_{(N)}^2]}{2} = FENR$$

Where again (N) = index of the donor cell.

The mass now in cell (L) is equal to  $\text{DELM} = \text{AMX}(L) + \text{GAMC}(j) + \text{AMMY} - \text{AMPY} - \text{AMMP} + \text{BMASS M}) - \text{FMASS}$  which equals the original mass plus the mass flow across the left, the bottom, behind, less the mass flow across the top, right and in front.

The total X momenta that has come into or left cell (L) is  $= \text{SIGMU} = \text{Fleft}(j) + \text{AMMU} + \text{BXMOM}(M) - \text{AMUT} - \text{AMUR} - \text{FXMOM} =$  the momenta crossing the left boundary plus the momenta crossing the bottom boundary plus the momenta crossing the back boundary less the momenta crossing the top, the right and front boundary.

The total Y momenta that has come into or left cell (L)  $= \text{SIGMV} = \text{YAMC}(j) + \text{AMMV} + \text{BYMOM}(M) - \text{AMVT} - \text{AMVR} - \text{FYMOM} =$  momenta crossing the left, the bottom and behind less the momenta crossing the top, right and front boundary.

The total Z momenta that has come into or left cell (L)  $= \text{SIGMW} = \text{ZMOM}(j) + \text{AMMW} + \text{BZMOM}(M) - \text{AMWT} - \text{AMWR} - \text{FZMOM} =$  momenta crossing the left, the bottom and behind less the momenta crossing the top, right and front boundary.

Now by conserving momenta and total energy, we can calculate new velocities and specific internal energy of cell (L)

$$\text{MU}_{LE} + \text{MU}_B + \text{MU}_{BH} - \text{MU}_T - \text{MU}_R - \text{MU}_F + \text{MU}_{(L)} = (\text{DELM}) U_{(L)}^{n+1}$$

and

$$\text{MV}_{LE} + \text{MV}_B + \text{MV}_{BH} - \text{MV}_T - \text{MV}_R - \text{MV}_F + \text{MV}_{(L)} = (\text{DELM}) V_{(L)}^{n+1}$$

and

$$\text{MW}_{LE} + \text{MW}_B + \text{MW}_{BH} - \text{MW}_T - \text{MW}_R - \text{MW}_F + \text{MW}_{(L)} = (\text{DELM}) W_{(L)}^{n+1}$$

and solve for the three velocities at cycle n+1.

The new specific internal energy is the total less the kinetic =

$$I_{(L)}^{n+1} = \frac{\text{E}_{LE} + \text{E}_B + \text{E}_{BH} - \text{E}_T - \text{E}_R - \text{E}_F + \text{E}_L}{\text{DELM}} - \frac{(U_{(L)}^{n+1})^2 + (V_{(L)}^{n+1})^2 + (W_{(L)}^{n+1})^2}{2}$$

DELM = the new mass. The subscripts LE, B, BH, T, R, F, L, refer to the left, bottom, behind, top, right, front and cell in question.

Now the five variables that were calculated for the top of cell (L) become the bottom quantities for cell (L + iMAX) and the variables that were calculated for the right of cell (L) become the left quantities for cell (L+1) and the variables that were calculated for the front of cell (L) become the back quantities for cell L + (iMAX)(jMAX). This completes the calculation for cell (L). After completion of PH2, all that remains to complete cycle n+1 is to update the pressures, which is done in the CDT routine.

### 3. TEST PROBLEMS

A series of test problems were undertaken to check out the TRIOIL code.

The early test problem consisted of 8 cells (2 in the x, 2 in the y and 2 in the z direction) containing internal energy only, free to expand into a vacuum. The purpose was to check the free surface treatment, the symmetry of disturbance in the three directions and the possible sphericity of the expansion at large distances. The results were encouraging, exact symmetry existed and the expansion was spherical considering the coarse zoning.

The second test problem (as shown in Fig. 3) consists of a grid 11 x 11 x 11. The corner cell has  $10^{16}$  ergs/g, surrounded by like material but cold. The three adjacent sides were treated as reflective boundaries, while the three opposite sides were transmittive boundaries. A check of computer results were made against the G. I. Taylor strong shock-point source solution. The comparison is presented in Fig. 4. As indicated, the position of the shock front is in good agreement with theory, however the magnitude is somewhat less in the TRIOIL solution. The latter results are consistent with similar calculations performed with the OIL (Ref. 1) code.

The third test problem was a normal impact to be compared with the OIL code. The configuration (as indicated in Fig. 5) for the TRIOIL code was a cube of 4 x 4 x 4 cells impacting normal to a semi-infinite target.

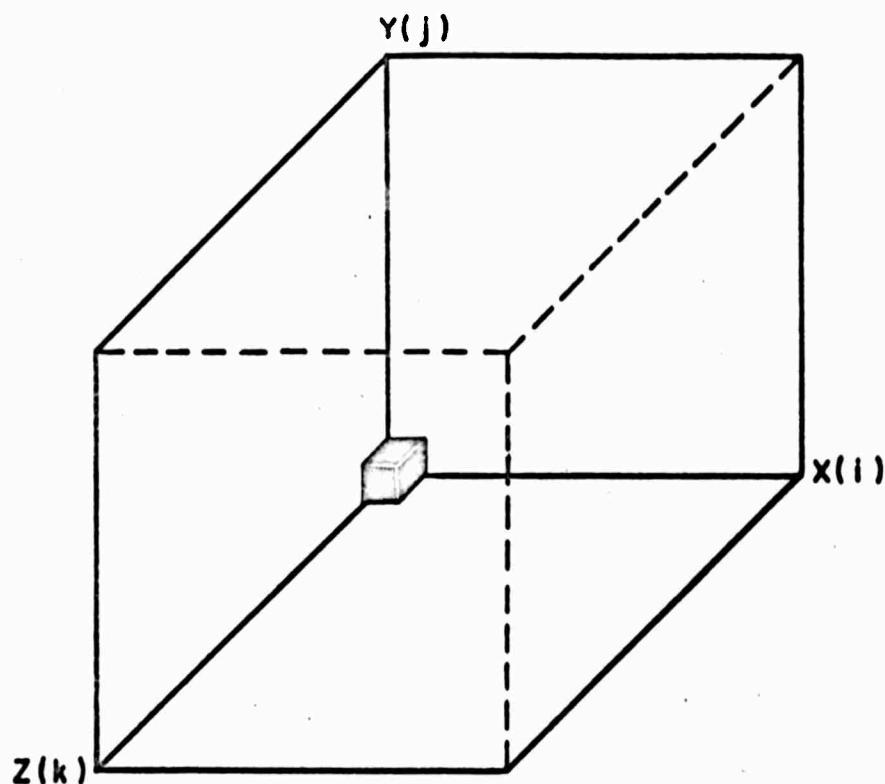
**TEST PROBLEM**

Fig. 3--The initial configuration for a point source calculation using the 3-D code for a 1-D problem

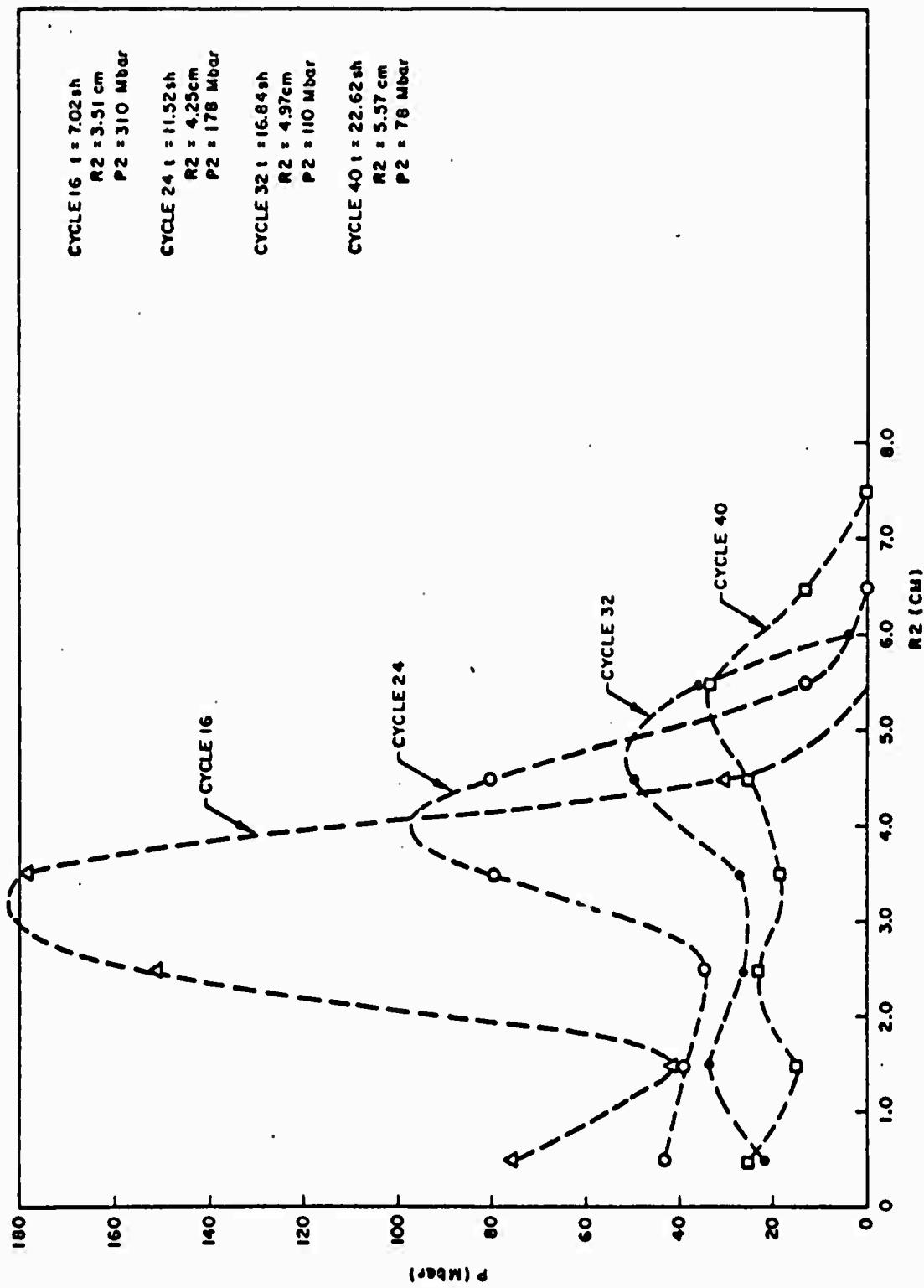


Fig. 4--Comparison of theory and computer results  
for a strong shock-point source solution

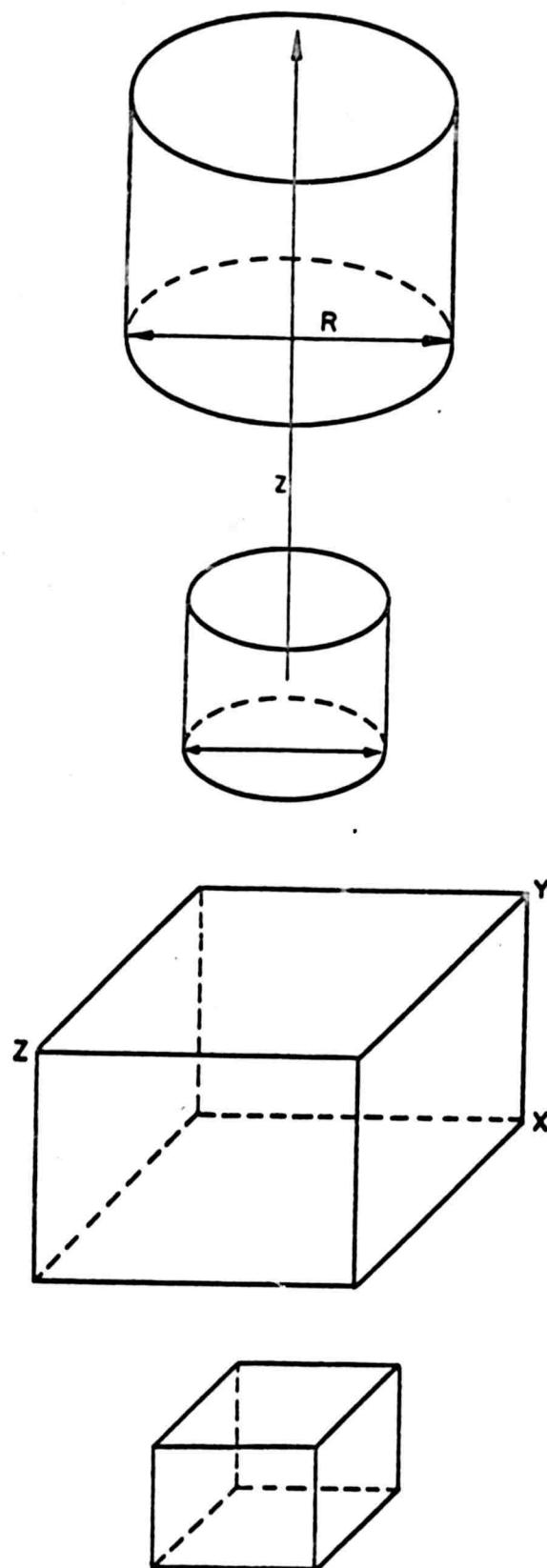


Fig. 5--Initial configuration of the OIL (top)  
and TRIOIL (bottom) problems

The configuration for OIL consisted of a right circular cylinder ( $D = L$ ) where the diameter was equal to the side of the cube in TRIOIL. The OIL problem had 2 radial zones and 4 axial in the projectile. The velocity of the projectile was  $2.6 \times 10^6$  cm/sec, and the projectile and target were both aluminum.

The momentum and times of the two problems were scaled by the ratio of their masses, the distances by the cube root of the ratio. Figure 6 presents the total positive y momentum (axial, in the case of OIL) versus time for the 2 codes. The agreement is excellent. Figure 7 displays the pressure attenuation into the target (along the axis for OIL and in either of the 2 inner original columns for the TRIOIL). Here again, we expect some difference at early times since one projectile is a cylinder and the other a cube, but the agreement is very good.

Figure 8 is a velocity (in the x-y plane) plot for the plane of symmetry,  $z = 0$ . Figure 9 is a velocity plot (in the y-z plane at a value of  $x$  in the center of the projectile. Exact symmetry is indicated by these two plots. Figure 10 is at a later time, presenting the velocity (in the x-y plane) plot for the plane of symmetry,  $z = 0$ .

Figure 11 is a velocity (in the x-y plane) plot for the plane of symmetry,  $z = 0$ , for an oblique impact ( $45^\circ$ ). The dark lines indicate the position of the original projectile to target at time  $t = 0$ . Figure 12 is a velocity plot in the y-z plane at a value of  $x$  at the right hand side of the original projectile.

Since the original formulation of the TRIOIL code, investigations into the effect of obliquity, and velocity, for semi-infinite and thin targets have been completed successfully.

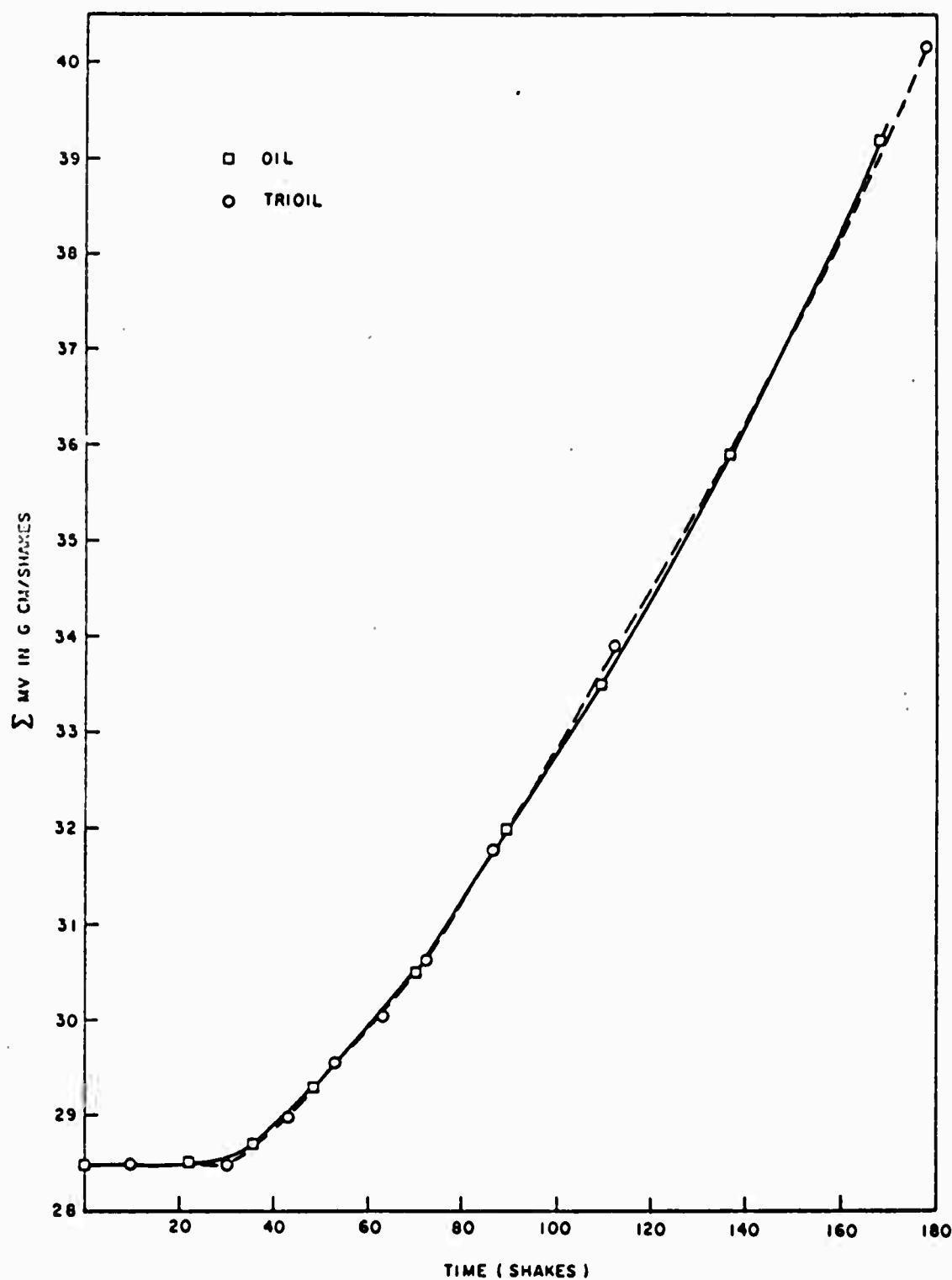


Fig. 6--Total positive y momentum as a function of time for the two codes

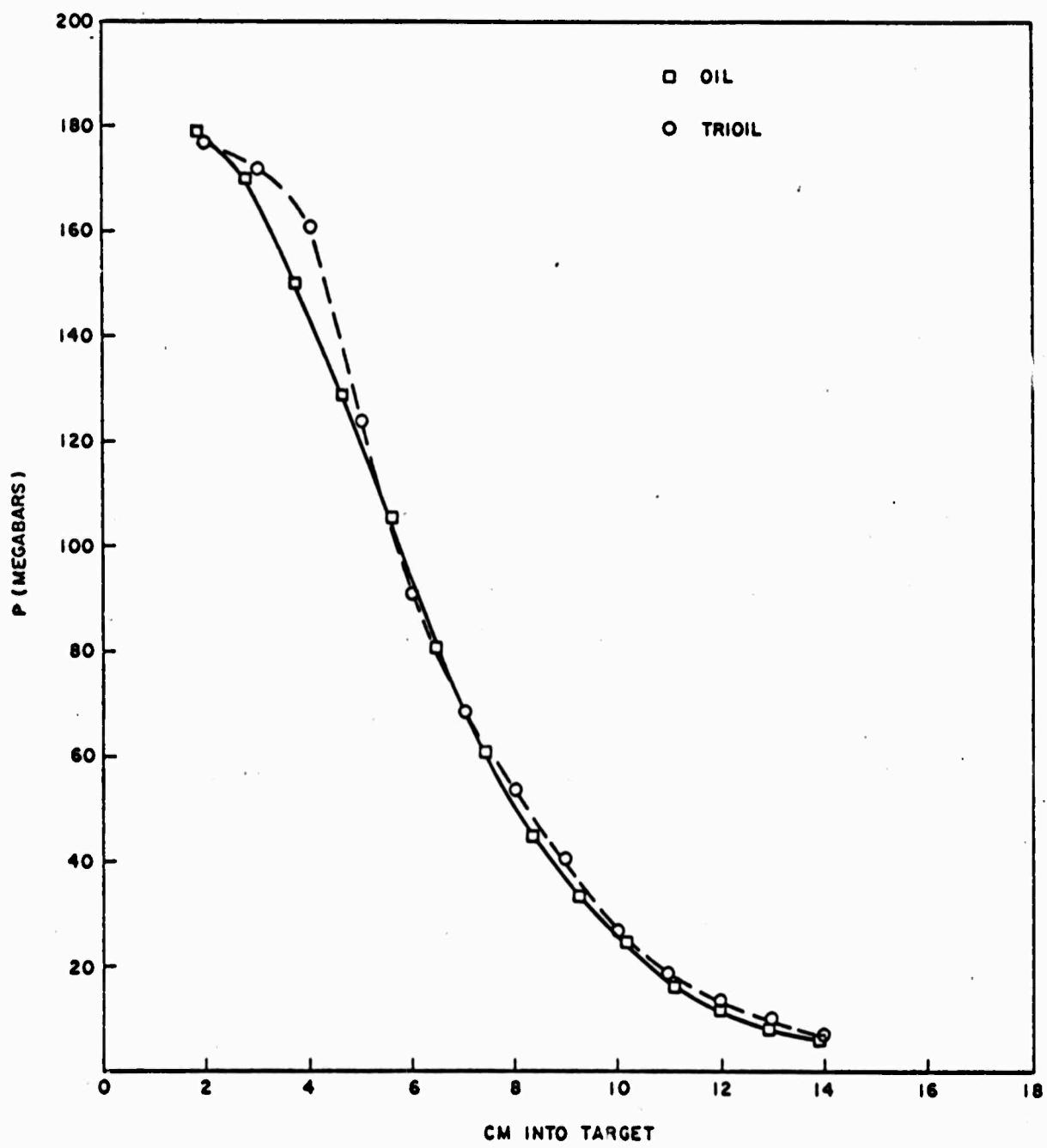


Fig. 7--Pressure attenuation into the target comparison for the 2 codes

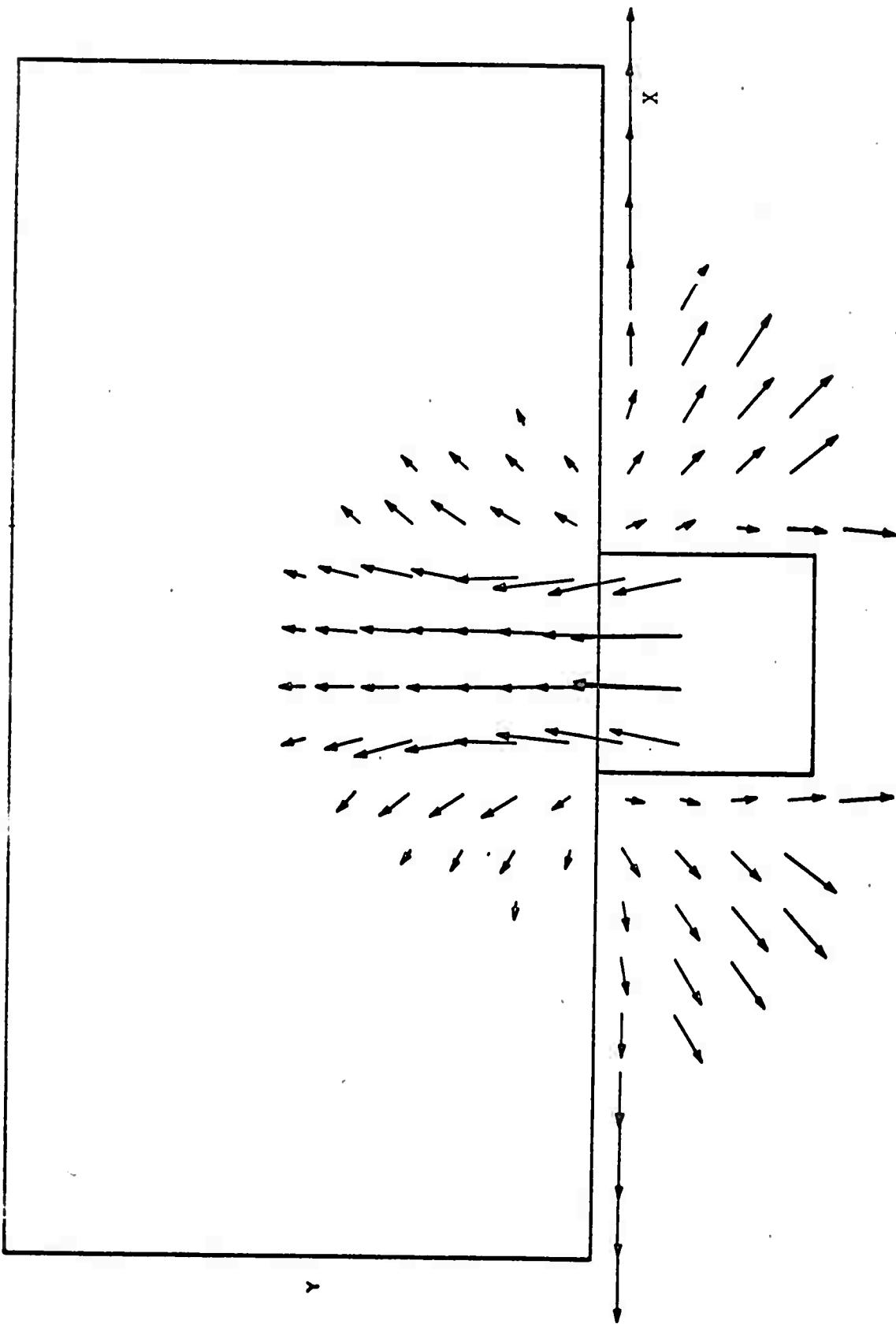


Fig. 8 - Velocities ( $X-Y$ ) in the plane of symmetry ( $Z=0$ ) for 90 degrees

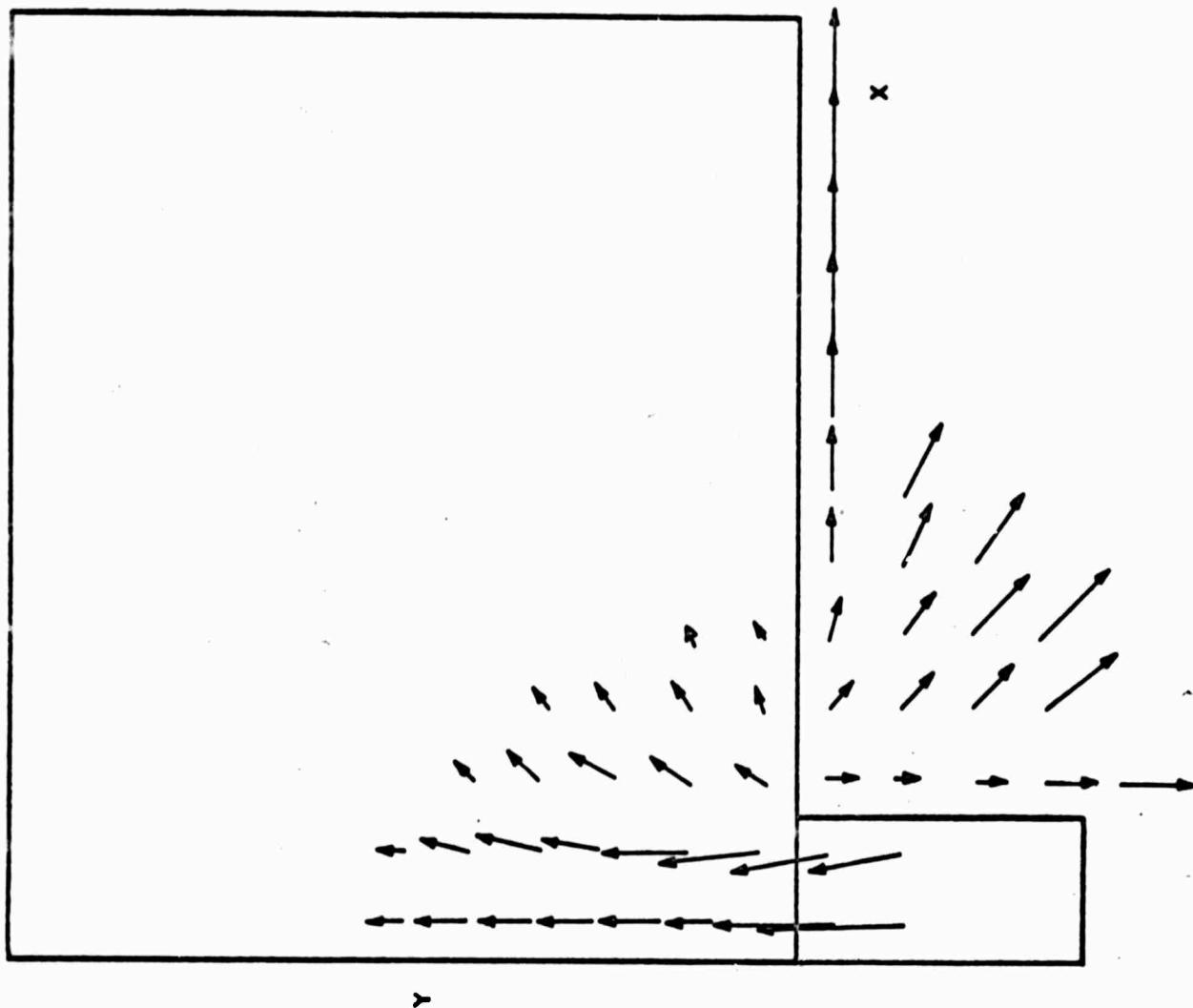


Fig. 9--Velocities (Y-Z) in the Y-Z plane at a X position  
of the original center of projectile for 20 degrees

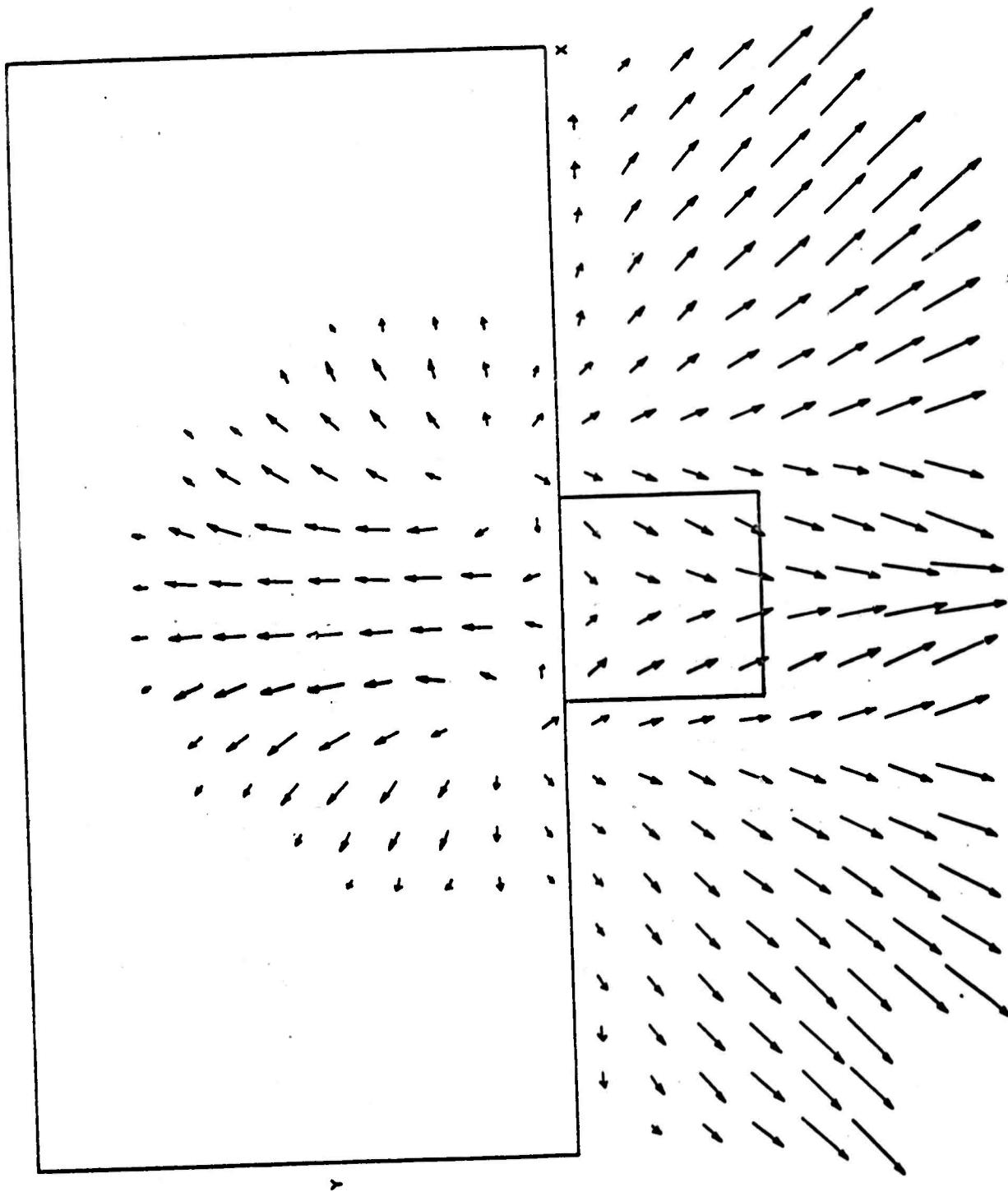


Fig. 10--Velocities (X-Y) in the plane  
of symmetry ( $Z=0$ ) for 90 degrees

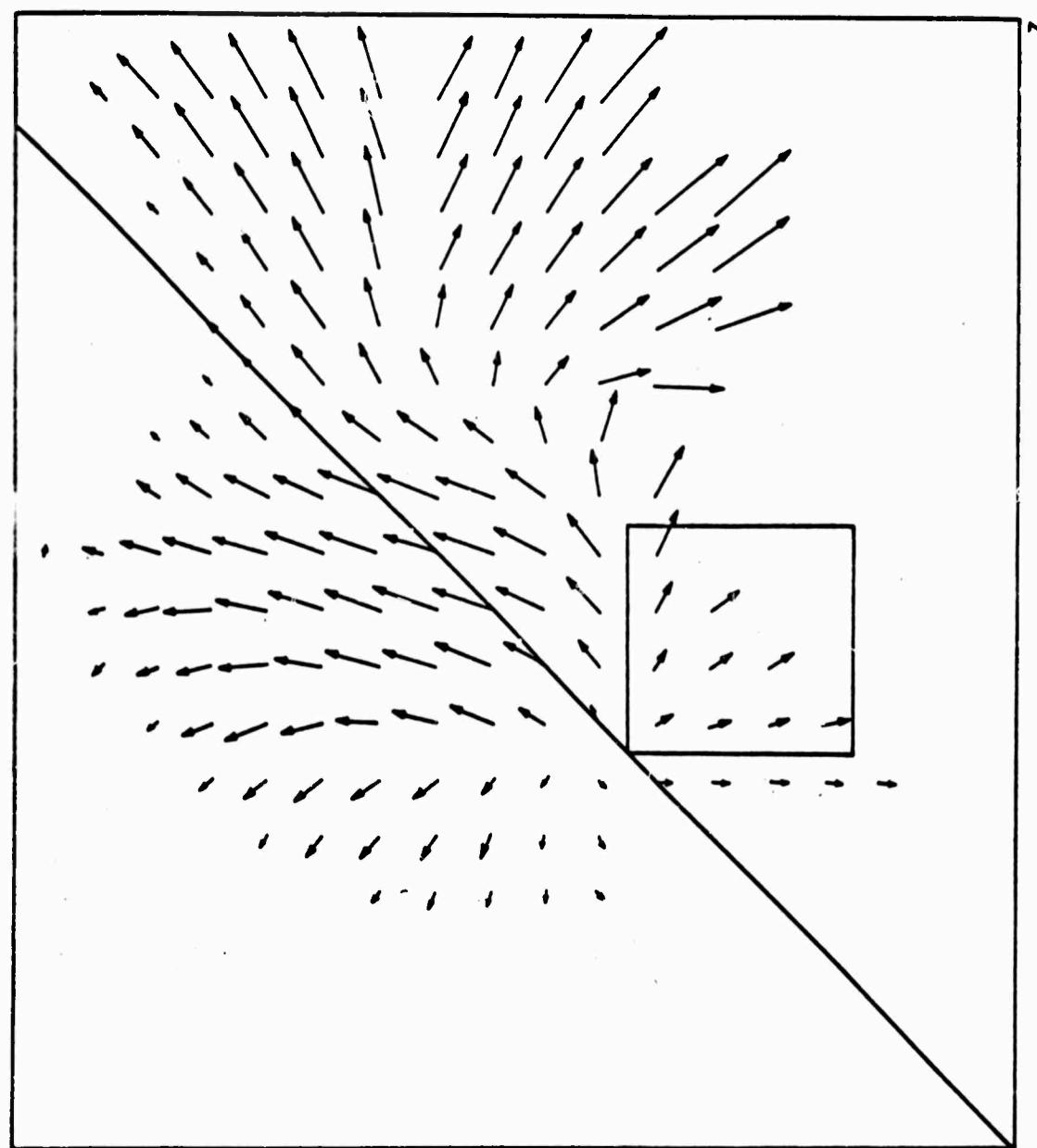


Fig. 11--Velocities (X-Y) in the plane of symmetry ( $Z=0$ ) for 45 degrees

1 WALLY

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C \*\*\*\* NOTE, THE FOLLOWING SET OF DIMENSION,  
C COMMON, AND EQUIVALENCE IS TO BE USED FOR ALL  
C SUBROUTINES WITH THE EXCEPTION OF SUBROUTINE CARDS.....

DIMENSION AIX(6000),AMX(6000),U(6000),V(6000),W(6000),P(6000),  
1DX(30),DY(30),DZ(30),UL(30),PL(30),X(30),Y(30),ZCOR(30),  
2Z(150),IZ(150),FLEFT(30),YAMC(30),SIGC(30),GAMC(30),ZMOM(30),  
3PBIND(700),UBIND(700),BMASS(700),BXMOM(700),BYMOM(700),  
4BZMOM(700),BENR(700)  
DIMENSION PR(50),PK(30)  
COMMON Z,AIX,AMX,U,V,W,P,DX,DY,DZ,  
1UL,PL,X,Y,ZCOR,PR,SIGC,GAMC,  
2ZMOM,UBIND,BMASS,BYMOM,  
3BZMOM,BENR,AREA,BIG,BOUNCE,PABOVE,PBLO,  
4PIDTS,PRR,RHO,SIG,UVMAX,VABOVE,VBL0,  
5VEL,WPS,WS,WSA,WSB,WSC,I,II,IN,IR,  
6IWSA,IWSB,IWSC,J,JN,JP,JR,K,KDT,KN,  
7KP,KR,KRM,L,M,MA,MB,MC,MD,ME,MZ,N  
COMMON REZ,TRAD,DTRAD,RADEB,RADER,RADET,X1,X2,Y1,Y2,IMAXA  
EQUIVALENCE (Z,IZ,PROB),(Z(2),CYCLE),(Z(3),DT),  
1(Z(4),PRINTS),(Z(5),PRINTL),(Z(6),DUMPT7),(Z(7),CSTOP),(Z(8),PIDY)  
2,(Z(9),GAM),(Z(10),GAMD),(Z(11),GAMX),(Z(12),ETH),(Z(13),FFA),  
3(Z(14),FFB),(Z(15),TMASS),(Z(16),XMAX),(Z(17),YMAX),(Z(18),ZMAX),  
4(Z(19),DNN),(Z(20),DMIN),(Z(21),DTNA),(Z(22),REZFCT),(Z(23),TOZONE)  
5),(Z(24),ECK),(Z(25),SBOUND),(Z(26),CABL0),(Z(27),T),(Z(28),GMAX),  
6(Z(29),WSGD),(Z(30),WSGX),(Z(31),GMADR),(Z(32),GMAXR),(Z(45),DTCHK)  
7),(Z(46),PCSTAB),(Z(47),CNOT),(Z(48),BFACT),(Z(49),EPSI),(Z(50),S1)  
8),(Z(51),S2),(Z(52),S3),(Z(53),S4),(Z(54),S5),(Z(55),S6),  
9(Z(56),S7),(Z(57),S8),(Z(58),S9),(Z(59),S10)  
EQUIVALENCE (Z(60),AMLOST),(Z(61),ELOST),(Z(62),XMLOST),  
1(Z(63),YMLOST),(Z(64),ZMLOST),(Z(65),ENEQ),(Z(66),RHONOT),  
2(Z(67),VELOC),(Z(68),BUG),(Z(81),NPR),(Z(82),NPRI),  
3(Z(83),NC),(Z(84),NPC),(Z(85),NRC),(Z(86),IMAX),(Z(87),JMAX),  
4(Z(88),KMAX),(Z(89),KMAXA),(Z(90),IXMAX),(Z(91),NOD),  
5(Z(92),NOPR),(Z(93),I1),(Z(94),I2),(Z(95),I3),(Z(96),I4),  
6(Z(97),N1),(Z(98),N2),(Z(99),N3),(Z(100),N4),(Z(101),N5),  
7(Z(102),N6),(Z(103),N7),(Z(104),N8),(Z(105),N9),(Z(106),N10),  
8(Z(107),N11),(Z(108),K1),(Z(109),K2),  
9(Z(110),J1),(Z(111),J2)  
EQUIVALENCE (BMASS,PBIND),(BXMOM,UBIND),  
1(UL,FLEFT),(PL,YAMC,PK)

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DI FOR MAIN/S,MAIN/S,MAIN/SS

INPU0760

C  
C \*\*\*\*\* 3DOIL \*\*\*\*\*

MAIN0050

C  
C THE INPUT ROUTINE WILL READ A TRIOIL BINARY DUMP  
C TAPE OR WILL CALL FOR SUBROUTINE SET-UP  
C WHICH WILL GENERATE A DATA TAPE  
C FOR RESTRICTED GEOMETRY

CALL INPUT

MAIN0060

```

C CDT ROUTINE CALLS FOR THE EQUATION OF STATE,
C CALCULATES THE DT(HYDRODYNAMICS TIME STEP) AND
C ADVANCES THE CYCLE NUMBER
10 CALL CDT MAIN0070

C IN EDIT , DETERMINE WHETHER TO EXECUTE A
C SHORT PRINT, LONG PRINT, BINARY TAPE DUMP
C OR TO STOP THE PROBLEM
CALL EDIT MAIN0080
*** SENSE LITE 1 SIGNIFIES THAT THIS
IS THE LAST CYCLE OF THIS RUN.
THE LITE IS TURNED ON IN THE EDIT ROUTINE.....
CALL SLITET(1,K000FX)
GO TO(30,20),K000FX MAIN0090
MAIN0100

C PH1 INTEGRATES THE MOMENTUM AND
C ENERGY EQUATIONS DUE TO
C TO PRESSURE FORCES ONLY , THE CONVECTIVE
C TERMS ARE TEMPORARILY DROPPED.
20 CALL PH1 MAIN0110

C PH2 SOLVES THE MASS TRANSPORT EQUATIONS
C AND MOVES MOMENTUM AND ENERGY
C ACROSS THE FIXED CELL BOUNDARIES
C TO APPROXIMATE THE CONVECTIVE TERMS THAT
C WERE OMITTED IN PH1
CALL PH2 MAIN0120
MAIN0130

C ALL CELL QUANTITIES , EXCEPT THE PRESSURE
C HAVE NOW BEEN ADVANCED TO
C CYCLE N+1
GO TO 10 MAIN0140
MAIN0150
30 CALL EXIT
END MAIN0170

NI FOR CARDS/S,CARDS/S,CARDS/SS
SUBROUTINE CARDS CARD0010
DIMENSION TABLE(1),CARD(7),LABLE(1) CARD0020
COMMON TABLE CARD0030
A 2 IN COLUMN 1, ROUTINE WILL FIX THE
FLOATING PT. NO.
A 1 IN COLUMN 1, MEANS THIS IS LAST CARD TO
READ IN.
EQUIVALENCE(TABLE(1),LABLE(1)) CARD0050
WRITE(6,10)

1 READ (5,11)IEND,LOC,NUMWPC,(CARD(I),I=1,NUMWPC)
WRITE(6,12)IEND,LOC,NUMWPC,(CARD(I),I=1,NUMWPC)
DO 4 I=1,NUMWPC CARD0080
J=LOC+I-1
IF(IEND-2)2,5,2 CARD0100
5 LABLE(J)=IFIX(CARD(I)) CARD0110
GO TO 4 CARD0120
2 TABLE(J)=CARD(I) CARD0130
4 CONTINUE CARD0140
IF(IEND-1)1,3,1 CARD0150
3 RETURN CARD0160
CARD0170
CARD0180

```

```

C FORMATS CARD0190
10 FORMAT(20H1 TR10IL INPUT CARDS///)
11 FORMAT(I1,I5,I1,0P7E9.4) CARD0210
12 FORMAT(1H I4,I7,I3,1P7E14.6) CARD0220
END CARD0230
@I FOR UNCLE/S,UNCLE/S,UNCLE/SS
SUBROUTINE UNCLE
REWIND N7
C SUBROUTINE (( UNCLE))) IS CALLED WHENEVER
C A CODED ERROR IS ENCOUNTERED
C IN ANY SUBROUTINE, ITS MAIN FUNCTION
C IS TO PRINT THE CELL QUANTITES OUT IN THE FORM
C OF THE NORMAL LONG PRINT
NR=90
WRITE(6,8120)NC
8120 FORMAT(1H0//68H AN ERROR HAS OCCURRED AND SUBROUTINE UNCLE HAS BEE
IN CALLED AT CYCLE15///)
5000 WRITE(6,8116)PROB,NC,T,DTNA,TRAD,DTRAD,NR,N1,N2,N3,N4
DO 1126 KK=K1,K2
WRITE(6,9041)KK,ZCOR(KK),DZ(KK)
5004 DO 5050 I=I1,I2
WS1=1.
J=J2+1
K=J2*I MAX+I+(KK-1)*IXMAX
DO 5046 L=J1,J2
J=J-1
K=K-I MAX
5012 IF(AMX(K))5046,5046,5014
5014 IF(WS1)5019,5019,5016
5016 WRITE(6,8135)I,X(I),DX(I)
WS1=0.
5019 WS=AMX(K)/(DX(I)*DY(J)*DZ(KK))
WSC=P(K)*1.E+4
5018 WRITE(6,8108)J,U(K),V(K),WSC,AMX(K),WS,AIX(K),W(K),Y(J)
5046 CONTINUE
5050 CONTINUE
1126 CONTINUE
RETURN
8108 FORMAT(I3,1X,1P8E12.5)
81160FORMAT(8H1PROBLEM6X,5HCYCLE9X,4HTIME13X,2HDT13X,4HTRAD11X,5HDTRAD1EDIT3380
12X,2HNR6X,2HN14X,2HN24X,2HN34X,2HN4/(F7.1,I11,2X,1P4E16.7,I10,2X,4EDIT3390
216))
81350FORMAT(1H //4H I =I3,6X,6HX(I) =F12.3,6X,7HDX(I) =F12.3//3H J8X,EDIT3520
11HX10X,1HY10X,3HF/A9X,3HAMX9X,3HRH08X,3HAIX9X,4H W 8X,2H Y/)
9041 FORMAT(1H //4H K =I3,6X,9HZCOR(K) =F12.3,6X,7HDZ(K) =F12.3)
END
@I FOR SETUP/S,SETUP/S,SETUP/SS
SUBROUTINE SETUP
C CALCULATE THE ADDITIONAL INDICES THAT ARE FUNCTIONS OF
C IMAX,JMAX, AND KMAX
C
IXMAX=(IMAX)*(JMAX)
KMAXA= KMAX*IXMAX
C
SET ALL CELL CENTERED QUANTITIES TO ZERO
C

```

```

DO 1 K=1,KMAXA
U(K)=0.
V(K)=0.
W(K)=0.
AIX(K)=0.
AMX(K)=0.
P(K)=0.
1 CONTINUE
X(1)=DX(1)

C
C      CALCULATE ALL X,S
DO 10 I=2,IMAX
X(I)=X(I-1)+DX(1)

C
C      NOTE, DX IS CONSTANT FOR ALL I
DX(I)=DX(1)
10 CONTINUE
Y(1)=DY(1)

C
C      CALCULATE ALL Y,S
DO 11 J=2,JMAX
Y(J)=Y(J-1)+DY(1)

C
C      NOTE, DY IS CONSTANT FOR ALL J
DY(J)=DY(1)
11 CONTINUE
ZCOR(1)=DZ(1)

C
C      CALCULATE ALL ZCOR,S
DO 12 K=2,KMAX
ZCOR(K)=ZCOR(K-1)+DZ(1)

C
C      NOTE, DZ IS CONSTANT FOR ALL K
DZ(K)=DZ(1)
12 CONTINUE

C
C      RHONOT IS INITIAL DESITY FOR ALL MATERIAL
J3=S1
DO 100 K=1,KMAX
LL=(K-1)*IXMAX
DO 100 I=1,IMAX
DO 100 J=J3,JMAX
L=LL+(J-1)*IMAX+I
AMX(L)=DX(I)*DY(J)*DZ(K)*RHONOT
100 CONTINUE

C
C      S1=INTERFACE(J) VALUE +1 BETWEEN PROJECTILE AND TARGET
C      S2=BACK BOUNDARY +1 OF THE PROJECTILE(K) VALUE
C      S3= FRONT BOUNDARY OF PROJECTILE(K) VALUE
C      S4= LEFT BOUNDARY(I) VALUE +1 OF THE PROJECTILE
C      S5= RIGHT BOUNDARY(I) OF THE PROJECTILE
C      S6= BOTTOM BOUNDARY (J) +1 OF THE PROJECTILE
C      S7= TOP BOUNDARY(J) OF THE PROJECTILE
ETH=0.
K11=S2
K22=S3
I11=S4

```

```

I22=S5
J11=S6
J22=S7
DO 200 K=K11,K22
LL=(K-1)*IXMAX
DO 200 I=I11,I22
DO 200 J=J11,J22
L=LL+(J-1)*IMAX+I
AMX(L)=DX(I)*DY(J)*DZ(K)*RHONOT
C   VELOC=INITIAL Y COMPONENT OF VELOCITY
V(L)=VELOC
C   S9= INITIAL X COMPONENT OF VELOCITY
U(L)=S9
C   S10= INITIAL Z COMPONENT OF VELOCITY
W(L)=S10
ETH=ETH+AMX(L)*(U(L)**2+V(L)**2+W(L)**2)/2.
200 CONTINUE
C
C   PRINT THE QUANTITIES ASSOCIATED WITH THE GRID
C
      WRITE(6,8000)IMAX,(X(I),I=1,IMAX)
      WRITE(6,8003)IMAX,(DX(I),I=1,IMAX)
      WRITE(6,8001)JMAX,(Y(J),J=1,JMAX)
      WRITE(6,8004)JMAX,(DY(J),J=1,JMAX)
      WRITE(6,8002)KMAX,(ZCOR(K),K=1,KMAX)
      WRITE(6,8005)KMAX,(DZ(K),K=1,KMAX)
      WRITE(6,8006)IMAX,JMAX,KMAX,IXMAX,KMAXA
      8000 FORMAT(1H /10H X(I) I=1,I2/(5F16.6))
      8001 FORMAT(1H /10H Y(J) J=1,I2/(5F16.6))
      8002 FORMAT(1H /13H ZCOR(K) K=1,I2/(5F16.6))
      8003 FORMAT(1H /11H DX(I) I=1,I2/(5F16.6))
      8004 FORMAT(1H /11H DY(J) J=1,I2/(5F16.6))
      8005 FORMAT(1H /11H DZ(K) K=1,I2/(5F16.6))
      8006 FORMAT(7I8)
C
C   WRITE A DUMP TAPE(FOR T=0.) FOR
C   THE TRIOIL CODE
REWIND N7
WS=555.0
WRITE(N7) WS,CYCLE,N3
WRITE(N7)(Z(I),I=1,150)
WRITE(N7)(U(I),V(I),W(I),AMX(I),AIX(I),I=1,KMAXA)
WRITE(N7)(X(I),I=1,IMAX)
WRITE(N7)(Y(J),J=1,JMAX)
WRITE(N7)(ZCOR(K),K=1,KMAX)
WS=666.
WRITE(N7) WS,WS,WS
RETURN
END
@I FOR INPUT/S,INPUT/S,INPUT/SS
SUBROUTINE INPUT
C
C
CALL SLITE (3)
IWSA=THE NUMBER OF BCD HEADER CARDS TO
READ IN
READ(5,8007)IWSA

```

INPU0010  
MAIN0020  
INPU0900  
INPU0980

```

DO 7 I=1,IWSA          INPU1000
READ (5,8004) IWS
WRITE(6,8004) IWS
7 CONTINUE
6 CALL CARDS          INPU1020
C     NOTE, PROVISIONS FOR CALLING A GENERATOR
C     CODE SUCH AS (CLAM), DOES NOT EXIST IN
C     3 DIMENSIONS AT THIS TIME
IF(PK(3))8887,8888,8888
8888 CALL CARDS
CALL SETUP
8887 CONTINUE          INPU1060
GO TO 1000             INPU1090
10 CONTINUE             INPU1120
CALL CARDS             INPU1130
GO TO 2000             INPU1140
40 DO 45 K=1,KMAXA    INPU1160
45 P(K)=0.0             INPU1170
C
C     INTEGRATE(THE TIME AND CYCLE NUMBER ) BACKWARDS, SINCE THEY
C     WILL BE ADVANCED IN CDT SUBROUTINE
C
T=T-UTNA               INPU1180
NC=NC-1                INPU1190
CYCLE=NC               INPU1200
NPC=NPC-1              INPU1210
UVMAX=0.0               INPU1220
C
C     CALCULATE DX,DY,DZ SINCE THEY ARE NOT ON THE TAPE
C
WS=0.
DO 50 I=1,IMAX          INPU1230
DX(I)=X(I)-WS
WS=X(I)
50 CONTINUE
WS=0.
DO 55 J=1,JMAX          INPU1250
DY(J)=Y(J)-WS
WS=Y(J)
55 CONTINUE
WS=0.
DO 65 K=1,KMAX
DZ(K)=ZCOR(K)-WS
WS=ZCOR(K)
65 CONTINUE
C DUMP THE Z BLOCK
K=1
DO 80 I=1,9
L=K+8
WRITE(6,8005)K,(Z(N),N=K,L)
80 K=L+1
K=81
DO 81 I=1,8
L=K+8
WRITE(6,8006)K,(IZ(N),N=K,L)
81 K=L+1

```

GO TO 10000 INPU1370  
 C  
 C  
 .C READ THE BINARY TAPE FOR A RESTART INPU1380  
 C  
 1000 MZ=150 INPU1390  
 IWS=0  
 1003 REWIND N7  
 1004 READ(N7)PR(1),PR(2),N3 INPU1400  
 NR=N3+6  
 1005 IF(PR(1)=555.0)1010,1016,1010 INPU1410  
 1010 I,S=IWS+1 INPU1420  
 1011 IF(MOD(IWS,3))9902,9902,1003  
 1016 IF(PR(2))1010,101c,1013  
 1018 IF(PR(2)=PR(2))1023,1023,1020  
 C READ OVER THIS IS NOT THE CORRECT CYCLE  
 1020 DO 1022 L=2,NR INPU1510  
 1022 READ(N7)WSS  
 GO TO 1004  
 1023 READ(N7)(Z(I),I=1,MZ)  
 C CHECK FOR CORRECT PROBLEM NUMBER  
 IF(ABS(PROB-PK(1))-0.01)1024,1024,9901 INPU1550  
 1024 READ(N7)(U(I),V(I),W(I),AMX(I),AIX(I),I=1,KMAXA)  
 READ(N7)(X(I),I=1,IMAX)  
 READ(N7)(Y(J),J=1,JMAX)  
 READ(N7)(ZCOR(K),K=1,KMAX)  
 1034 READ(N7)PR(1),PR(2),PR(3)  
 1036 IF(PR(1)=555.0)9904,1040,1038 INPU1680  
 1038 IF(PR(2)=666.0)9905,1040,9905 INPU1690  
 1040 GO TO 10 INPU1700  
 2000 IF(WSGX)9906,2010,2005 INPU1750  
 C CALCULATE 1./(GAMMA-1.) AND GAMMA/(GAMMA-1.)  
 2005 GAMX=1.0/(WSGX-1.0) INPU1760  
 2010 WSGX=(GAMX+1.0)/GAMX INPU1770  
 GMAXR=GAMX\*WSGX INPU1780  
 2012 IF(WSGD)9907,2020,2015 INPU1790  
 2015 GAMD=1.0/(WSGD-1.0) INPU1800  
 2020 WSGD=(GAMD+1.0)/GAMD INPU1810  
 GMADR=GAMD\*WSGD INPU1820  
 GMAX=WSGD INPU1830  
 C SEARCH FOR MAX GAMMA  
 IF(WSGD>WSGX)2025,2030,2030 INPU1840  
 2025 GMAX=WSGX INPU1850  
 2030 GO TO 40 INPU1860  
 C ERROR INPU1900  
 9901 NK=1023 INPU1910  
 GO TO 9999 INPU1920  
 9902 NK=1011 INPU1930  
 GO TO 9999 INPU1940  
 9904 NK=1036 INPU1950  
 GO TO 9999 INPU1960  
 9905 NK=1038 INPU1970  
 GO TO 9999 INPU1980  
 9906 NK=2000 INPU1990  
 GO TO 9999 INPU2000  
 9907 NK=2012 INPU2010  
 9999 NR=1 INPU2020

```

      WRITE(6,8000)PR(1),PR(2),PK(1),PK(2),PK(3),IWS
      CALL UNCLE
      CALL DUMP
C.
      10000 RETURN          INPU2040
C.                               INPU2050
C.                               INPU2060
C.                               INPU2070
C.   FORMATS
      8000 FORMAT(1P5E14.6,I5)    INPU2090
      80040FORMAT(I1,71H)         INPU2100
      1           )
      8005 FORMAT(I4,1X,1P9E12.5)
      8006 FORMAT(I4,1X,9I7)
      8007 FORMAT(6I3)
C.
      END                  INPU2120
      01 FOR   CDT/S,CDT/S,CDT/SS    INPU2130
          SUBROUTINE CDT          CDT 0010
C.
      3000 VEL=0.0          CDT 0020
C.
C.   SET UP THE LOOPS FOR CALCULATING THE MATERIAL PRESSURE
C.
      DO 3050 K=K1,K2
      LL=(K-1)*IXMAX
      3005 DO 3050 I=I1,I2
      3010 L=I+LL+(J1-1)*IMAX
      3015 DO 3050 J=J1,J2
      3020 IF(AMX(L))9901,3050,3025
C.
C.   THE ES ROUTINE CALCULATES THE PRESSURE, FOR INPUT IT NEEDS I,J,K,L
C.   AND THE AIX(ENERGY) AND AMX(MASS)
C.
      3025 CALL ES              CDT 1110
C.
C.   THE OUTPUT FROM ES IS THE P(PRESSURE) AND MAX (GAMMA-1.)
C.
      3030 IF(ABS(P(L))-1.E-20)3035,3035,3040
      3035 P(L)=0.
      3040 IF(WSGX-VEL)3050,3050,3045          CDT 1150
      3045 VEL=WSGX               CDT 1160
      3050 L=L+IMAX
      3055 KDT=1                 CDT 1180
      UVMAX=-1.0                CDT 1190
C.
C.   NOW SET UP THE LOOP FOR CALCULATING DT FROM THE PARTICLE
C.   VELOCITIES AND THE COURANT CONDITION
C.
C.   WE FLAG THE CELL THAT IS CONTROLLING THE TIME STEP
C.   AND STORE THE VALUES OF I,J,K INTO
C.   N10,N11, AND N9 RESPECTIVELY.
C.
      DO 3255 K=K1,K2
      :   LL=(K-1)*IXMAX
      3070 DO 3255 I=I1,I2
      3075 L=I+LL+(J1-1)*IMAX
      3095 DO 3255 J=J1,J2
      3100 CONTINUE

```

IF(AMX(L))9901,3255,4  
 C IF RHO IS LESS THAN DTCHK, OMIT THE STABILITY CHECK FOR THIS CELL  
 4 IF(AMX(L)/(DX(I)+DY(J)\*DZ(K))-DTCHK)3255,3255,3115  
 3115 SIG=DX(I) CDT 1260  
 3120 IF(DY(J)-SIG)3120,3130,3130 CDT 1270  
 3125 SIG=DY(J)  
 3130 IF(DZ(K)-SIG)5131,5130,5130 CDT 1280  
 5131 SIG=DZ(K)  
 C HERE WE CALCULATE THE SPEED OF SOUND , THE PERFECT GAS SPEED OF  
 C SOUND OR DR/DHO FROM THE METAL EQUATION OF STATE  
 5130 IF(C,0,T)4000,4000,4001  
 4000 WS=SQRT(OMAX\*P(L)/AMX(L)+DX(I)+DY(J)\*DZ(K))  
 GO TO 3205 CDT 1310  
 4001 WSA=ABS(P(L))\*1.E+4  
 WSA=NOT+3FACT+(WSA\*\*EPSI)  
 WSA=WAI.E-3  
 3205 WS=WS/SIG  
 3210 IF(UVMAX-WS)3215,3220,3220 CDT 1340  
 3215 N10=1 CDT 1350  
 N11=0 CDT 1360  
 N9=K CDT 1370  
 UVMAX=WS CDT 1380  
 3220 CONTINUE CDT 1390  
 2 WS=ABS(U(L))/DX(I)  
 3225 IF(UVMAX-WS)3230,3235,3235 CDT 1450  
 3230 UVMAX=WS CDT 1460  
 N10=1 CDT 1470  
 N9=K  
 N11=J CDT 1480  
 3235 WS=ABS(V(L))/DY(J)  
 3240 IF(UVMAX-WS)3245,3250,3250 CDT 1500  
 3245 N10=1 CDT 1510  
 N11=J CDT 1520  
 N9=K  
 UVMAX=WS CDT 1530  
 3250 CONTINUE CDT 1540  
 WS=ABS(W(L))/DZ(K)  
 IF(UVMAX-WS)5245,5250,5250  
 5245 N9=K  
 N10=1  
 N11=J  
 UVMAX=WS  
 5250 CONTINUE  
 3255 L=L+1,MAX  
 IF(UVMAX)9912,9912,3260  
 C  
 C HERE,CHECK THE 3 POSSIBLE OPTIONS FOR CALCULATING THE  
 C NEW DT  
 C  
 6260 IF(CAULN)90-91,3300 CDT 1560  
 C HERE OPTIONS EXIST FOR STARTING THE PROBLEM  
 C WITH SMALL TIME STEPS(A SMALL FRACTION OF STABILITY)  
 C FOR PROBLEMS WHERE THE INITIAL ENERGY IS  
 C PRIMARILY INTERNAL.....  
 90 IF(Z(69)=1.0)93,94,94  
 93 Z(69)=Z(69)+Z(70)  
 GO TO 95

```

94 Z(69)=1.0
95 DT=.5/VEL/UVMAX*PCSTAB*Z(69)
   GO TO 3295                                CDT 1580
   . . 91 WS=UVMAX*DT
   . . WSA=0.5/VEL                            CDT 1590
   . . 3265 IF(FFA-WSA)3276,3276,3270        CDT 1600
   . . 3270 FFA=WSA                            CDT 1610
   . . 3276 IF(WS-FFA)3285,3300,3280        CDT 1620
   . . 3280 DT=DT/WS*FFB/0.9                  CDT 1630
   . . GO TO 3295                            CDT 1640
   . . 3285 IF(WS-FFB)3290,3290,3300        CDT 1650
   . . 3290 DT=DT*FFA/WS*0.9                  CDT 1660
   . . 3295 KDT=0                            CDT 1670
   . . 3300 T=T+DTNA                          CDT 1680
   . . 82 NC=NC+1                           CDT 1690
   . . CYCLE=NC                           CDT 1700
   . . UVMAX=UVMAX*DT                      CDT 1710
   . . NPC=NPC+1                           CDT 1720
   . . 3305 IF(T)9909,3320,3310            CDT 1730
   . . 3310 IF(KDT)9910,3315,3320          CDT 1740
   . . 3315 WRITE(6,8000)T,DTNA,DT          CDT 1750
   . . 3320 DTNA=DT                           CDT 1760
   . . GO TO 3325                          CDT 1770
   . . 9901 NK=3020                           CDT 1780
   . . GO TO 9999                           CDT 1790
   . . 9909 NK=3305                           CDT 1800
   . . GO TO 9999                           CDT 1810
   . . 9910 NK=3310                           CDT 1820
   . . GO TO 9999                           CDT 1830
   . . 9912 NK=1                            CDT 1840
   . . GO TO 9999                           CDT 1850
   . . 9999 NR=2                            CDT 1860
   . . WRITE(6,8002)I,J,K,L,NK,NR,NC
   . . WRITE(6,8001)UVMAX,AMX(L),P(L),DT,VEL
   . . 8001 FORMAT(1P5E14.6)
   . . 8002 FORMAT(8I5)
   . . CALL UNCLE
   . . CALL DUMP
   . . 3325 RETURN                         CDT 1870
   . . 8000 FORMAT (17H0CHANGE DT ... T=1PE10.3,10H    DT(N)=1PE10.3,12H      DT
   . . 1(N+1)=1PE10.3)
   . . END                               CDT 1880
   . . QI FOR PH1/S,PH1/S,PH1/SS
   . . SUBROUTINE PH1                         PH1 0010
C
C THE VELOCITIES,ENERGIES,PRESSURE AND MASS ARE AT
C CELL CENTERS
C FIRST PASS ,CALCULATE U,V AND W FOR CYCLE N+1 ,AND THE WORK
C TERMS USING THE VELOCITIES AT CYCLE (N).
C SECOND PASS, CALCULATE THE CONTRIBUTION TO THE CHANGE IN
C INTERNAL ENERGY FROM WORK TERMS EVALUATED FROM U,V AND W
C AT CYCLE N+1
C 2 PASSES REQUIRED
C
C REMEMBER, WE ARE NOT ADVANCING THE VELOCITIES AND ENERGY TO CYCLE
C N+1,SINCE WE HAVE NOT AS YET SOLVED THE TRANSPORT TERMS, AS
C USUAL, WE REFER TO THE VELOCITIES AND ENERGY FROM THIS ROUTINE

```

```

C AS TILDA QUANTITIES
C
C CELL IN QUESTION = L=(J-1)*IMAX+I+(K-1)*IMAX*JMAX
C
C IXMAX=IMAX*JMAX
C
C KMAXA=(IMAX)(JMAX)(KMAX)
C
C CELL TO THE RIGHT= L+1
C
C CELL ABOVE =N= L+IMAX
C
C CELL IN FRONT =NN =L+IXMAX
C
C N1 =FLAG AT THE LEFT
C N2 =FLAG AT THE RIGHT
C N3 =FLAG AT THE TOP
C N4 =FLAG AT THE BOTTOM
C N5 =FLAG AT THE BEHIND
C N6 =FLAG AT THE IN FRONT
C
C SET THE FLAGS FOR INCREASING OR DECREASING ACTIVE GRID
C COUNTERS TO ZERO
C IX1=0
C IX2=0
C JY1=0
C JY2=0
C KZ1=0
C KZ2=0
C SET FLAG FOR SUBCYCLING
C VEL=1.
C
C RETURN HERE FOR THE SECOND PASS
C
C SET UP THE K LOOP FIRST
2 DO 3 K=K1,K2
7 L=(K-1)*(IXMAX)+1
C
C CHECK FOR BOUNDARY CONDITIONS AT THE LEFT
C
8 DO 3302 J=1,JMAX
6 IF(N1)9,99,9
C
C TRANSMITTIVE LEFT BOUNDARY
C
99 PL(J)=0.
UL(J)=U(L)
GO TO 10
C
C REFLECTIVE LEFT BOUNDARY
C
9 PL(J)=P(L)
UL(J)=0.
10 L=L+IMAX
3302 CONTINUE
11 IF(K-1)8999,12,7001

```

7001 IF(K-K1)15,12,15  
C  
C BACK SIDE BC. HAVE ALREADY BEEN SET  
C  
C  
12 CONTINUE  
C BACK SIDE IS TRANS, BUT WILL TAKE CARE OF IT LATER  
C BACK SIDE IS REFLECTIVE  
C  
13 DO 2302 N=1,IXMAX  
IF(N5)6010,6011,6010  
6011 PBIND(N)=0.  
UBIND(N)=W(N)  
GO TO 2302  
6010 PBIND(N)=P(N)  
UBIND(N)=0.  
2302 CONTINUE  
15 LL=(K-1)\*IXMAX  
C  
C SET DO LOOP IN X DIRECTION  
C  
DO 4 I=I1,I2  
16 M=(J1-1)\*IMAX+I  
17 L=LL+I+(J1-1)\*IMAX  
C  
C SET DO LOOP IN Y DIRECTION  
C  
18 DO 5 J=J1,J2  
NN=L+IXMAX  
19 N=L+IMAX  
20 IF(J-1)9902,21,7003  
7003 IF(J1-J)3305,23,3305  
C  
C WE HAVE ALREADY CALCULATED THE BOTTOM BC.  
C CHECK FOR BOTTOM BOUNDARY  
21 IF(N4)22,23,22  
C  
C BOTTOM BOUNDARY IS TRANS  
C  
23 VBL0=V(L)  
PBL0=0.  
GO TO 3305  
C  
C BOTTOM BOUNDARY IS REFLECTIVE  
C  
22 VBL0=0.  
PBL0=P(L)  
C  
C NOW,WE HAVE THE LEFT BC.(IF REFLECTIVE)AND THE BOTTOM(IF REFLECT)  
C  
3305 IF(AMX(L))9900,3340,3306  
C  
C CELL IN QUESTION IS VOID, GET OUT AND CONTINUE THE LOOP  
C  
3306 IF(IMAX-I)9901,3311,3307  
C  
C WE ARE AT THE RIGHT BOUNDARY OF THE GRID

C  
C        THE TOP IS REFLECTIVE  
C  
30 PABOVE=P(L)  
VABOVE=0.  
GO TO 3328  
C  
C        THE TOP IS TRANS  
C  
31 PABOVE=PBLO  
C  
MODIFY ETH FOR TRANS BOUNDARY  
C  
ETH=LTH-PABOVE/2.\*V(L)+DT\*DX(I)\*DZ(K)  
GO TO 3323  
C  
C        WE ARE NOT AT THE TOP  
3320 IF(ANX(N))9905,3322,3324  
C  
C        CELL ABOVE IS VOID  
3322 PABOVE=0.  
3323 VABOVE=V(L)  
GO TO 3328  
C  
C        NORMAL FLOW FOR ALL CELLS OCCUPIED  
3324 PABOVE=(P(L)+P(N))/2.  
32 IF(1-J)3325,33,9906  
C  
C        BOTTOM BOUNDARY HAS BEEN SET  
C        WE ARE AT THE BOTTOM  
33 IF(N4)3325,7000,3325  
C  
C        REFLECTIVE BOTTOM BOUNDARY CONDITION HAS ALREADY  
C        BEEN SET.  
C  
7000 PBLO=PABOVE  
LTH=LTH+PBLO/2.\*V(L)+DT\*DX(I)\*DZ(K)  
3325 VABOVE=(V(L)+V(N))/2.  
C  
C        CHECK THE Z DIRECTION  
C  
3328 IF(KMAX-K)9907,4418,4420  
C  
C        WE ARE AT THE FRONT OF THE GRID  
C  
4418 IF(N6)2999,34,2999  
C  
C        FRONT IS REFLECTIVE  
2999 PZR=P(L)  
WZR=0.  
GO TO 4328

```

C      FRONT IS TRANS
C      34 PZR=PBIND(M)
C
C      MODIFY ETH FOR TRANS BOUNDARY
C      4419 ETH=LTH-PZR/2.*W(L)*DT*DX(I)*DY(J)
C          GO TO 4323
C
C      CHECK CELL IN FRONT
C
C      4420 IF(AMX(NN))9908,35,4324
C
C      CELL IN FRONT IS OCCUPIED
C      CELL IN FRONT IS VOID
C
C      35 PZR=0.
C      4323 WZR=W(L)
C          GO TO 4328
C
C      NORMAL FLOW IN Z DIRECTION FOR ALL CELLS OCCUPIED
C      4324 PZR=(P(L)+P(NN))/2.
C          IF(1-K)4325,37,9909
C
C      BC. BEHIND HAVE ALREADY BEEN SET
C
C      WE ARE IN THE FIRST (X-Y )PLANE K=1
C      37 IF(N5)4325,8000,4325
C
C      REFLECTIVE BC. BEHIND HAVE ALREADY BEEN SET
C      TRANS BC. IN THE BACK
C      8000 PBIND(M)=PZR
C
C      MODIFY ETH FOR TRANS BOUNDARY
C      ETH=LTH-PZR/2.*W(L)*DT*DY(J)*DX(I)
C      4325 WZR=(W(L)+W(NN))/2.
C
C      CHECK FOR FIRST OR SECOND PASS
C
C      4328 IF(VLL)9910,42,3400
C
C      THIS IS THE SECOND PASS, SKIP THE MOMENTA EQUATIONS
C
C
C      INTEGRATE THE Y COMPONENT OF VELOCITY (V)
C      3400 V(L)=V(L)+(PULO-P ABOVE)/AMX(L)*DT*DX(I)*DZ(K)

```

C        CHECK THE MASS TO THE RIGHT  
 3307 IF(AMX(L+1))9903,3312,3314  
 C  
 C        THIS IS THE BC. AT THE RIGHT OF A OCCUPIED CELL ,WITH THE  
 C        NEIGHBOR VOID.  
 C  
 3312 PRR=0.  
 3313 URRE=U(L)  
 GO TO 3316  
 C  
 C        HERE, WE ARE AT THE RIGHT BOUNDARY OF GRID (I=IMAX)  
 C        CHECK HERE FOR REFLECT OR TRANS BC.  
 C  
 3311 IF(N2)3308,3309,3308  
 C  
 C        REFLECTIVE  
 C  
 3308 PRR=P(L)  
 URRE=0.  
 GO TO 3316  
 C  
 C        TRANSMITTIVE  
 C  
 3309 PRR=PL(J)  
 C  
 C        MODIFY ETH HERE AT THE TRANS BOUNDARY  
 C  
 ETHE=ETH-PRR/2.\*U(L)\*DT\*DY(J)\*DZ(K)  
 GO TO 3313  
 C  
 C        HERE IS NORMAL FLOW FOR ALL CELLS OCCUPIED  
 3314 PRR=(P(L)+P(L+1))/2.  
 URRE=(U(L)+U(L+1))/2.  
 C  
 C        CHECK HERL FOR ALONG THE LEFT BOUNDARY (I=1) FOR TRANS  
 3316 IF(I-1)9911,50,3310  
 50 IF(N1)3310,51,3310  
 C  
 C        REFLECTIVE (BUT BC. HAVE ALREADY BEEN SET)  
 C  
 C        TRANSMITTIVE  
 51 PL(J)=PRR  
 C  
 C        MODIFY ETH AT TRANS BOUNDARY  
 52 ETHE=ETH+PRR/2.\*U(L)\*DT\*DY(J)\*DZ(K)  
 GO TO 3310  
 3310 IF(JMAX-J)9904,3318,3320  
 C  
 C        WE ARE AT THE TOP OF THE GRID  
 3318 IF(N3)30,31,30

```

IF(ABS(V(L))-1.E-8)3401,3401,3402
3401 V(L)=0.

C
C      INTEGRATE THE X COMPONENT OF VELOCITY (U)
3402 U(L)=U(L)+(PL(J)-PRR)/AMX(L)*DT*DY(J)*DZ(K)
    40 IF(ABS(U(L))-1.E-8)3403,3403,3404
3403 U(L)=0.

C
C      INTEGRATE THE Z COMPONENT OF VELOCITY (W)
3404 W(L)=W(L)+(PBIND(M)-PZR)/AMX(L)*DT*DY(J)*DX(I)
    41 IF(ABS(W(L))-1.E-8)3405,3405,42
3405 W(L)=0.

C
C
C      HERE CALCULATE THE CHANGE IN INTERNAL ENERGY DUE TO THE
C      WORK TERMS
42 WS=P(L)*DT/AMX(L)*((VBL0-VABOVE)/2.*DX(I)*DZ(K)
    1+(UL(J)-URR)/2.*DY(J)*DZ(K)
    2+(UBIND(M)-WZR)/2.*DX(I)*DY(J))
43 AIX(L)=AIX(L)+WS

C
C      CHECK ON ADVANCING OR DECREASING GRID COUNTERS
C
5600 IF(I-I2)5999,5801,5801
5801 IF(IX2)5999,5802,5999
5802 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5803,5999,5803
5803 IX2=1
    GO TO 5999
5999 IF(K-K2)4999,5804,5804
5804 IF(KZ2)4999,5805,4999
5805 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5806,4999,5806
5806 KZ2=1
    GO TO 4999
4999 IF(KZ1)5300,5222,5300
5222 IF(K1-1)5300,5300,5000
5000 IF(K1-K)5300,5001,5001
5001 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5002,5300,5002
5002 KZ1=1
5300 IF(JY1)5600,5304,5600
5304 IF(J1-1)5600,5600,5301
5301 IF(J1-J)5600,5302,5302
5302 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5303,5600,5303
5303 JY1=1
5600 IF(IX1)3342,5604,3342
5604 IF(I1-1)3342,3342,5601
5601 IF(I1-I)3342,5602,5602
5602 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5603,3342,5603
5603 IX1=1
    GO TO 3342

C
C      CAME HERE BECAUSE THE CELL IN QUESTION (L) IS VOID
C
3340 PRR=0.
    URR=U(L+1)
    PABOVE=0.
    VABOVE=V(N)
    PZR=0.

```

WZR=W(NN)  
C  
C SET THE ABOVE QUANTITIES TO BELOW  
• 3342 VBLO=VABOVE  
PBLO=PABOVE  
C  
• C SET THE RIGHT QUANTITIES TO THE LEFT  
PL(J)=PRR  
UL(J)=URR  
C  
C SET THE FRONT QUANTITIES TO BEHIND  
PBIND(M)=PZR  
UBIND(M)=WZR  
C  
C UPDATE THE INDICES  
C  
L=N  
M= M+IMAX  
C TERMINATION OF LOOP ON J---(Y)  
5 CONTINUE  
C CHECK ON ADVANCING OR DECREASING GRID COUNTERS  
5700 LJ=L-IMAX  
IF(JY2)4,5701,4  
5701 IF(ABS(U(LJ))+ABS(V(LJ))+ABS(W(LJ))+AIX(LJ))5702,4,5702  
• 5702 JY2=1  
GO TO 4  
C  
• C TERMINATION OF LOOP ON I---(X)  
4 CONTINUE  
C  
C TERMINATION OF LOOP ON K---(Z)  
3 CONTINUE  
C  
C  
C CHECK FOR FIRST OR SECOND PASS  
44 IF(VEL-1.)46,45,46  
45 VEL=0.  
C  
C RECYCLE  
GO TO 2  
C  
C HAVE COMPLETED BOTH PASSES  
46 CONTINUE  
C  
C INCREASE OR DECREASE COUNTERS AS REQUIRED  
C  
I1=I1-IX1  
5900 IF(I1)5901,5901,5902  
5901 I1=1  
5902 I2=I2+IX2  
5903 IF(I2-IMAX)5905,5905,5904  
5904 I2=IMAX  
5905 J1=J1-JY1  
IF(J1)5906,5906,5907  
5906 J1=1  
5907 J2=J2+JY2  
IF(J2-JMAX)5909,5909,5908

```

5908 J2=JMAX
5909 K1=K1-KZ1
IF(K1)5910,5910,5911
5910 K1=1
5911 K2=K2+KZ2
IF(K2-KMAX)5913,5913,5912
5912 K2=KMAX
5913 RETURN
8999 NK=11
GO TO 9999
9902 NK=20
GO TO 9999
9900 NK=3305
GO TO 9999
9901 NK=3306
GO TO 9999
9903 NK=3307
GO TO 9999
9911 NK=3316
GO TO 9999
9904 NK=3310
GO TO 9999
9905 NK=3320
GO TO 9999
9906 NK=32
GO TO 9999
9907 NK=3328
GO TO 9999
9908 NK=4420
GO TO 9999
9909 NK=4324
GO TO 9999
9910 NK=4328
9999 NR=3
WRITE(6,8500)I,J,K,L,M,N,NN,NK,NR
8500 FORMAT(9I6)
CALL UNCLE
C CALL DUMP
C CALL DUMP
END
@I FOR PH2/S,PH2/S,PH2/SS
SUBROUTINE PH2
DIMENSION AIX(6000),AMX(6000),U(6000),V(6000),W(6000),P(6000),
C **** 3DOIL ****
C HERE, WE APPROXIMATE THE TRANSPORT TERMS LEFT OUT OF THE
C MOMENTUM AND ENERGY EQUATIONS IN PH1 BY MOVING
C MASS, (SOLVING THE MASS CONSERVATION EQUATION) THIS MASS
C THEN CARRIES ENERGY AND MOMENTUM ACROSS THE FIXED
C GRID LINES
C AMPY = MASS FLOW AT THE TOP
C AMUT = X MOMENTUM COMPONENT OF THIS MASS
C AMVT = Y MOMENTUM COMPONENT OF THIS MASS
C AMWT = Z MOMENTUM COMPONENT OF THIS MASS
C DELET = SPECIFIC ENERGY OF THIS MASS

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PH2 0010

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C AMRP = MASS FLOW AT THE RIGHT
C AMUR = X MOMENTUM COMPONENT OF THIS MASS
C AMVR = Y MOMENTUM COMPONENT OF THIS MASS
C AMWR = Z MOMENTUM COMPONENT OF THIS MASS
C DELER= SPECIFIC ENERGY OF THIS MASS
C
C AMNY= MASS FLOW AT THE BOTTOM
C AMMU= X MOMENTUM COMPONENT OF THIS MASS
C AMMV= Y MOMENTUM COMPONENT OF THIS MASS
C AMMW= Z MOMENTUM COMPONENT OF THIS MASS
C DELEB= SPECIFIC ENERGY OF THIS MASS
C
C GAMC = MASS FLOW AT THE LEFT
C FLEFT= X MOMENTUM COMPONENT OF THIS MASS
C YAMC = Y MOMENTUM COMPONENT OF THIS MASS
C ZMOM = W MOMENTUM COMPONENT OF THIS MASS
C SIGC = SPECIFIC ENERGY OF THIS MASS
C
C BMASS= MASS FLOW AT THE BACK
C BXMOM= X MOMENTUM COMPONENT OF THIS MASS
C BYMOM= Y MOMENTUM COMPONENT OF THIS MASS
C BZMOM= Z MOMENTUM COMPONENT OF THIS MASS
C BENR = SPECIFIC ENERGY OF THIS MASS
C
C FMASS= MASS FLOW IN FRONT
C FXMOM= X MOMENTUM COMPONENT OF THIS MASS
C FYMOM= Y MOMENTUM COMPONENT OF THIS MASS
C FZMOM= Z MOMENTUM COMPONENT OF THIS MASS
C FENR = SPECIFIC ENERGY OF THIS MASS
C
C REZ=0.
C
C INITIALIZE THE FLAGS FOR ADVANCING THE
C ACTIVE GRID COUNTERS TO ZERO
C IX1=0
C IX2=0
C JY1=0
C JY2=0
C KZ1=0
C KZ2=0
C SUM=0.
C CALL SLITE(0)
C
C DO LOOP ON K
C
C 5 DO 1 K=K1,K2
C     LL=(K-1)*IXMAX
C
C     DO LOOP ON I
C 6 DO 2 I=I1,I2
C
C     NOTE (L) IS THE CELL INDEX =(J-1)IMAX + I
C           + (K-1)IXMAX (NOTE IXMAX=(IMAX)(JMAX))
C
C     L=LL+I+(J1-1)*IMAX
C

```

C DO LOOP ON J

C

7 DO 3 J=J1,J2  
 NN = THE INDEX OF THE CELL IN FRONT = L+ IXMAX  
 NN=L+IXMAX

C

N= THE INDEX OF THE CELL ABOVE = L+ IMAX  
 N=L+IMAX

C

M= INDEX OF THE CELL IN QUESTION FOR A SINGLE PLANE

C

M=(J-1)\*IMAX+I  
 N1,N2,N3,N4,N5,N6 ARE FLAGS TO SET BOUNDARY CONDITIONS  
 AT THE 6 FACES OF THIS GRID

C

N1 REFERS TO THE LEFT  
 N2 REFERS TO THE RIGHT  
 N3 REFERS TO THE TOP  
 N4 REFERS TO THE BOTTOM  
 N5 REFERS TO BEHIND  
 N6 REFERS TO IN FRONT

C

FREE SURFACES WITHIN THE GRID ARE TREATED AS FOLLOWS ,  
 IF THE MASS FLOW INTO A EMPTY CELL PRODUCES A  
 DENSITY THAT IS LESS THAN TOZONE(A INPUT NUMBER LIKE  
 .001 OF RHONOT),THE MASS FLUX IS SET TO ZERO

C

600 IF(J-1)9903,601,9302  
 9302 IF(J1-J)9,603,9

C

BOTTOM BC. HAS BEEN SET

C

601 IF(AMX(L))9904,603,602

C

WE ARE AT THE BOTTOM OF THE X-Y PLANE  
 602 IF(V(L))604,603,603

C

SET Y COMPONENT OF MOMENTUM TO 0.  
 603 AMMV=0.  
 GO TO 698

C

CALCULATE THE MASS FLUX AT THE BOTTOM  
 604 AMMY=AMX(L)/DY(J)\*V(L)\*DT

C

CHECK SO MASS FLUX DOES NOT MORE THAN EMPTY THE CELL  
 605 IF(AMMY+AMX(L))9300,607,607  
 9300 AMMY=-AMX(L)  
 607 IF(N4)609,608,609

C

BOTTOM BOUNDARY IS TRANS  
 608 AMMU=AMMY\*U(L)

C

CALCULATE THE 3 MOMENTAS,THE ENERGY  
 SUBTRACT THIS ENERGY LOSS FROM ETH  
 AMMV=AMMY\*V(L)  
 AMMW=AMMY\*W(L)  
 WS=(U(L)\*\*2+V(L)\*\*2+W(L)\*\*2)/2.

```

DELEB=AIX(L)+WS
ETH=ETH+AMMY*DELEB
IF(-AMMY/(DX(I)*DY(J)*DZ(K))-Z(80))610,610,6600
6600 REZ=1.0
GO TO 610
C
C BOTTOM BOUNDARY IS REFLECTIVE, NET MOMENTA CHANGE= 2MV
609 AMMV=2.*AMMY*V(L)
C
C SET MASS, X AND Z MOMENTA AND SPECIFIC ENERGY TO 0.
698 AMMY=0.
AMMU=0.
AMMW=0.
DELEB=0.
610 CONTINUE
C
C *** FINISHED WITH THE BOTTOM BC. ****
C ****
9 IF(I-1)8999,10,9301
9301 IF(I1-I)506,9310,506
9310 JJ=J
GO TO 20
C
C WE ARE ALONG THE LEFT BOUNDARY ( I=1)
10 NL=L
11 JJ=J
15 IF(AMX(NL))9900,20,16
20 FLEFT(JJ)=0.
GO TO 5504
16 IF(U(NL))17,20,20
C
C CALCULATE MASS FLUX
17 GAMC(JJ)=AMX(NL)/DX(1)*U(NL)*DT
21 IF(GAMC(JJ)+AMX(NL))22,500,500
22 GAMC(JJ)=-AMX(NL)
500 IF(N1)501,502,501
C
C LEFT BOUNDARY IS TRANS
C CALCULATE THE 3 MOMENTAS, THE ENERGY
C SUBTRACT THIS ENERGY LOSS FROM ETH
502 FLEFT(JJ)=GAMC(JJ)*U(NL)
YAMC(JJ)=GAMC(JJ)*V(NL)
ZMOM(JJ)=GAMC(JJ)*W(NL)
WS=(U(NL)**2+V(NL)**2+W(NL)**2)/2.
SIGC(JJ)=AIX(NL)+WS
ETH=ETH+GAMC(JJ)*SIGC(JJ)
IF(-GAMC(JJ)/(DX(I)*DY(J)*DZ(K))-Z(75))503,503,6610
6610 REZ=1.0
GO TO 503
C
C LEFT BOUNDARY IS REFLECTIVE, NET MOMENTA CHANGE =2MU
501 FLEFT(JJ)=2.*GAMC(JJ)*U(NL)
C
C SET MASS, Y AND Z MOMENTA AND SPECIFIC ENERGY TO 0.
5504 GAMC(JJ)=0.
YAMC(JJ)=0.
ZMOM(JJ)=0.

```

SIGC(JJ)=0.  
 503 CONTINUE  
 103 CONTINUE  
 C:  
 C: FINISHED WITH LEFT BOUNDARY CONDITIONS  
 C:  
 C: \*\*\*\*  
 C:  
 506 IF(K-1)9901,23,9303  
 9303 IF(K1-K)31,250,31  
 23 IF(AMX(L))9902,250,24  
 250 BZMOM(M)=0.  
 GO TO 25  
 C:  
 C: CHECK Z COMPONENT OF VELOCITY  
 24 IF(W(L))26,250,250  
 C:  
 C: SET THE 5 DATA BEHIND TO 0.  
 25 BMASS(M)=0.  
 BXMOM(M)=0.  
 BYMOM(M)=0.  
 BENR(M)=0.  
 GO TO 31  
 C:  
 C: VELOCITY IS - , CALCULATE THE MASS FLUX  
 26 BMASS(M)=AMX(L)/DZ(K)\*W(L)\*DT  
 C:  
 C: CHECK SO WE DONT EMPTY MORE MASS THAN THERE IS  
 27 IF(BMASS(M)+AMX(L))28,40,40  
 28 BMASS(M)=-AMX(L)  
 C:  
 C: CHECK FOR TRANS OR REFLECT  
 40 IF(N5)41,29,41  
 C:  
 C: REFLECTIVE  
 41 BZMOM(M)=2.\*BMASS(M)\*W(L)  
 GO TO 25  
 C:  
 C: TRANSMITTIVE  
 C: CALCULATE THE MOMENTA OF THIS MASS  
 29 BXMOM(M)=BMASS(M)\*U(L)  
 BYMOM(M)=BMASS(M)\*V(L)  
 BZMOM(M)=BMASS(M)\*W(L)  
 C:  
 C: CALCULATE THE TOTAL ENERGY CARRIED BY THIS MASS  
 30 WS=(U(L)\*\*2+V(L)\*\*2+W(L)\*\*2)/2.  
 BENR(M)=AIX(L)+WS  
 C:  
 C: REMOVE THE ENERGY LOSS FROM ETH  
 ETH=ETH+BMASS(M)\*BENR(M)  
 IF(-BMASS(M)/(DX(I)\*DY(J)\*DZ(K))-Z(78))31,31,6620  
 6620 REZ=1.0  
 C:  
 C: HAVE CALCULATED THE DATA BEHIND, NOW CHECK ON JMAX  
 C:  
 C: \*\*\*\*  
 C: NOW, UP TO THIS POINT, WE HAVE TAKEN CARE OF

C BOTH REFLECTIVE AND TRANSMITTIVE BOUNDARIES  
C AT THE BOTTOM, LEFT, AND BACK  
C \*\*\*\*  
C \*\*\*\*  
C  
C VABOVE CALC.  
C \*\*\*\*  
31 VEL=0.  
C  
C SET UP TO CALCULATE VABOVE  
IF(JMAX-J)211,211,207  
C  
C WE ARE AT THE TOP OF THE GRID  
211 VEL=1.  
GO TO 208  
C  
C CHECK CELL ABOVE  
207 IF(AMX(N))215,215,214  
C  
C CELL ABOVE IS VOID  
214 IF(AMX(L))216,216,209  
C  
C CELL (L) IS VOID, BUT CELL ABOVE IS OCCUPIED  
216 VABOVE=V(N)  
GO TO 212  
215 IF(AMX(L))205,205,208  
C  
C BOTH CELLS ARE VOID  
C  
205 VABOVE=0.  
GO TO 212  
208 VABOVE=V(L)  
GO TO 212  
C  
C BOTH CELLS ARE OCCUPIED  
209 VABOVE=(V(L)+V(N))/2.  
212 FS=0.  
C  
C U RIGHT CALC.  
C \*\*\*\*  
C  
C NOW,BEGIN CALCULATION OF URR  
C  
404 IF(IMAX-I)412,412,405  
405 IF(AMX(L+1))411,411,409  
409 IF(AMX(L))410,410,407  
C  
C CELL (L) IS VOID ,BUT CELL TO THE RIGHT IS FILLED  
410 URR=U(L+1)  
GO TO 408  
C  
C MASS ON THE RIGHT=0.  
411 IF(AMX(L))403,403,406  
403 URR=0.

C GO TO 408  
 C WE ARE AT THE RIGHT OF THE GRID  
 412 FS=1.  
 406 URR=U(L)  
 GO TO 408  
 C THIS IS THE NORMAL PATH  
 C BOTH CELLS ARE FILLED  
 407 URR=(U(L)+U(L+1))/2.  
 408 CONTINUE  
 C  
 C W IN FRONT CALC.  
 C \*\*\*\*\*  
 C NOW ,WE HAVE VABOVE (V AT THE TOP) AND URR(V AT THE RIGHT)  
 5503 AREA=0.  
 C LETS CALCULATE WOUT (THE Z COMPONENT)  
 504 IF(KMAX-K)512,512,505  
 505 IF(AMX(NN))511,511,509  
 509 IF(AMX(L))510,510,507  
 C CELL (L) IS VOID ,BUT CELL IN FRONT IS OCCUPIED  
 510 WOUT=W(NN)  
 GO TO 508  
 C MASS IN FRONT IS 0.  
 511 IF(AMX(L))5513,5513,5066  
 5513 WOUT=0.  
 GO TO 508  
 C WE ARE AT THE FRONT BOUNDARY  
 512 AREA=1.  
 5066 WOUT=W(L)  
 GO TO 508  
 C NORMAL PATH  
 C BOTH CELLS ARE FILLED  
 507 WOUT=(W(L)+W(NN))/2.  
 C  
 C IF VEL IS GREATER THAN 0. ,WE ARE IN THE TOP CELL (J=JMAX)  
 C IF FS IS GREATER THAN 0., WE ARE IN THE RIGHT COLUMN  
 C THAT IS (I=IMAX)  
 C IF AREA IS GREATER THAN 0., WE ARE IN THE FRONT SLAB(X-YPLANE)  
 C THAT IS (K=KMAX)  
 508 CONTINUE  
 C  
 C \*\*\*\*\*  
 C NOW, FINALLY, WE HAVE ALL 3 INTERFACE VELOCITIES  
 C \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \*  
 C HERE WE BEGIN THE CALCULATION OF THE 3 FLUXES  
 C  
 C \*\*\*\*\*  
 C IF TOP IS TRANS. THE ENERGY LOSS IS CALCULATED LATER  
 C \*\*\*\*\*  
 100 IF(VABOVE)102,101,1103

```

C
C      Y FLUX IS POSITIVE
1103 IF(AMX(L))9910,101,104
.C
.C      SET INDICES
104 IF(J-1)9910,6104,6105
.C 6105 KP=L-IMAX
     IF(AMX(KP))9910,6106,6104
6106 IF(ABS(VABOVE-VELOC)/VELOC-BUG)107,6104,6104
6104 LY=L
     JY=J
     IF(VEL)105,105,109
C
C      Y FLUX = 0.
101 AMPY=0.
     AMUT=0.
     AMVT=0.
     AMWT=0.
     DELET=0.
     GO TO 115
C
C      Y FLUX IS NEGATIVE (DOWN)
C      FLUX IS NEGATIVE FROM CELL (L)
102 IF(VEL)106,106,101
.C
C      FLUX IS - BUT FROM THE TOP CELL
C      CHECK CELL ABOVE
106 IF(AMX(N))9911,101,107
C
C      FLUX IS NEGATIVE, BUT CELL MASS =0.
107 LY=N
     JY=J+1
105 VABOVE= (VABOVE)/(1.+(V(N)-V(L))/DY(JY)*DT/SBOUND)
C
C      CALCULATE FLUX AT THE TOP
C      * * * * *
109 AMPY=AMX(LY)*VABOVE/DY(JY)*DT
110 IF(VEL)115,115,111
C
C      WE ARE AT THE TOP, CHECK THE BC. AT THE TOP
111 IF(N3)112,115,112
C
C      REFLECT THE MASS
112 IF(AMPY)115,115,113
113 AMVT=-2.*AMPY*V(L)
114 AMPY=0.
     AMUT=0.
     AMWT=0.
     DELET=0.
*****IF RIGHT IS TRANS. THE ENERGY LOSS IS CALCULATED LATER.*****
C      BEGIN CALCULATING THE FLUX AT THE RIGHT
115 IF(URR)118,116,117
116 AMMP=0.
     AMUR=0.

```

AMVR=0.  
AMWR=0.  
DELER=0.

C C C  
CALCULATE THE MASSFLOW IN THE Z DIRECTION  
GO TO 300

117 IF(AMX(L))9911,116,120  
120 LX=L  
IX=I  
IF(FS)119,119,122

C C C  
RIGHT FLUX IS NEGATIVE  
118 IF(FS)130,130,116

C C C  
FLUX IS NEGATIVE ,BUT CELL MASS IS ZERO ALSO

C C C  
CHECK THE CELL TO THE RIGHT  
130 IF(AMX(L+1))9911,116,121  
121 LX=L+1  
IX=I+1  
119 WS=(U(L+1)-U(L))/DX(IX)\*DT/SBOUND  
URR=URR/(1.+WS)

C C C  
CALCULATE THE MASS FLUX AT THE RIGHT  
\* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \* \*  
122 AMMP=AMX(LX)/DX(IX)\*URR\*DT  
IF(FS)123,300,123

C C C  
CHECK THE BOUNDARY CONDITION  
123 IF(N2)124,300,124

C C C  
TRANS  
124 IF(AMMP)300,300,125

C C C  
REFLECT THE MASS  
125 AMUR=-2.\*AMMP\*U(L)  
AMMP=0.  
AMVR=0.  
AMWR=0.  
DELER=0.

C C C  
IF FRONT IS TRANS. THE ENERGY LOSS IS TAKEN CARE OF LATER.

C C C  
DO THE Z COMPONENT NOW

C C C  
SET THE 5 DATA IN FRONT TO 0.  
300 IF(WOUT)318,316,317  
316 FMASS=0.  
FXMOM=0.  
FYMOM=0.  
FZMOM=0.  
FENR=0.  
GO TO 700  
317 IF(AMX(L))9912,316,3200  
3200 LZ=L  
IZZ=K

```

C IF(AREA)319,319,322
C FRONT FLUX IS NEGATIVE
318 IF(AREA)320,320,316
C
C FLUX IS NEGATIVE, BUT FROM IN FRONT
C CHECK CELL IN FRONT
320 IF(AMX(NN))9912,316,321
C
C FLUX IS NEGATIVE ,BUT CELL MASS =0.
321 LZ=NN
IZZ=K+1
319 WS=(W(NN)-W(L))/DZ(IZZ)*DT/SBOUND
WOUT=WOUT/(1.+WS)
C
C CALCULATE THE MASS FLUX IN FRONT
C * * * * * * * * * * * *
322 FMASS=AMX(LZ)/DZ(IZZ)*WOUT*DT
IF(AREA)323,700,323
323 IF(N6)324,700,324
C
C TRANS
C
C REFLECT,WE ARE IN FRONT
324 IF(FMASS)700,700,325
C
C REFLECT THE MASS
325 FZMOM=-2.*FMASS*W(L)
FMASS=0.
FXMOM=0.
FYMOM=0.
FENR=0.
C
C *****
C NOW WE HAVE ALL 3 FLUXES AND ALL
C THE BOUNDARY CONDITIONS HAVE BEEN SET
700 IF(AMPY)760,980,761
C
C TOP FLUX IS -
760 IF(AMPY+AMX(N))762,980,980
762 AMPY=-AMX(N)
GO TO 980
761 IF(-AMPY+AMX(L))763,980,980
763 AMPY=AMX(L)
980 IF(AMMP)7300,981,7301
7300 IF(AMMP+AMX(L+1))7302,981,981
7302 AMMP=-AMX(L+1)
GO TO 981
7301 IF(-AMMP+AMX(L))7303,981,981
7303 AMMP=AMX(L)
981 IF(FMASS)7400,982,7401
7400 IF(FMASS+AMX(NN))7402,982,982
7402 FMASS=-AMX(NN)
GO TO 982
7401 IF(-FMASS+AMX(L))7403,982,982
7403 FMASS=AMX(L)
982 WS=GAMC(J)

```

IF(WS)902,901,901  
901 WS=0.  
.902 WSA=AMMY  
IF(WSA)904,903,903  
903 WSA=0.  
.904 WSB=BMASS(M)  
IF(WSB)906,905,905  
905 WSB=0.  
906 WSC=AMX(L)+WS+WSA+WSB  
907 WS=AMPY  
IF(WS)950,950,909  
950 WS1=0.  
IF(K-K1)951,953,951  
951 IF(GAMC(J+1))952,953,953  
952 WS1=GAMC(J+1)  
953 WS2=0.  
NA=M+IMAX  
IF(K-K1)955,957,955  
955 IF(BMASS(NA))956,957,957  
956 WS2=BMASS(NA)  
957 WS3=WS1+WS2+AMX(N)  
958 IF(AMPY+WS3)959,908,908  
959 AMPY=-WS3  
GO TO 908  
908 WS=0.  
909 WSA=AMMP  
IF(WSA)970,970,911  
970 WS1=0.  
IF(K-K1)971,973,971  
971 IF(BMASS(M+1))972,973,973  
972 WS1=BMASS(M+1)  
973 WS3=WS1+AMX(L+1)  
974 IF(AMMP+WS3)975,910,910  
975 AMMP=-WS3  
GO TO 910  
910 WSA=0.  
911 WSB=FMASS  
IF(WSB)912,912,913  
912 WSB=0.  
913 WST=WS+WSA+WSB  
IF(WST)921,921,931  
931 IF(WSC)932,932,933  
932 IF(AMPY)934,934,935  
935 AMPY=0.  
934 IF(AMMP)936,936,937  
937 AMMP=0.  
936 IF(FMASS)921,921,938.  
938 FMASS=0.  
GO TO 921  
933 IF(WSC-WST)914,921,921  
914 WSD=WSC/WST  
WS=WS\*WSD  
WSA=WSA\*WSD  
WSB=WSB\*WSD  
915 IF(WS)917,917,916  
916 AMPY=WS  
917 IF(WSA)919,919,918

```

918 AMMP=WSA
919 IF(WSB)921,921,920
920 FMASS=WSB
921 CONTINUE
    IF(AMPY)703,2700,,02
2700 IF(J-JMAX)717,2701,2701
2701 IF(N3)716,717,716
    717 AMUT=0.
    AMVT=0.
    AMWT=0.
    DELET=0.
    GO TO 716
C
C      TOP FLUX IS +
702 IF(JMAX-J)9913,704,705
C
C      CHECK CELL ABOVE
705 IF(AMX(N))9914,706,704
C
C      FREE SURFACE AT TOP
706 IF(AMPY/(DX(I)*DY(J)*DZ(K))-TOZONE)707,704,704
C
C      DENSITY IS TOO SMALL, SET FLUX =0.
707 AMPY=0.
    GO TO 717
C
C      TOP FLUX IS NEGATIVE
703 IF(J-1)9913,701,709
709 IF(AMX(L))9914,710,701
710 IF(-AMPY/(DX(I)*DY(J)*DZ(K))-TOZONE)711,701,701
711 AMPY=0.
    GO TO 717
C
C      ADD UP THE MASSES (REMEMBER THEY HAVE DIRECTION)
C
704 DELM=GAMC(J)+AMMY+BMASS(M)-AMPY
    IF(VEL)9914,712,720
C      AT TOP OF GRID
720 IF(N3)713,714,713
C
C      TOP BOUNDARY IS TRANSMITTIVE
714 WS= U(L)**2+V(L)**2+W(L)**2
    ETH=ETH-AMPY*(AIX(L)+WS/2.)
    IF(AMPY/(DX(I)*DY(J)*DZ(K))-Z(79))712,712,6630
6630 REZ=1.
C
C      CALCULATE THE MOMENTUMS
712 AMUT=AMPY*U(L)
    AMVT=AMPY*V(L)
    AMWT=AMPY*W(L)
    GO TO 713
701 AMUT=AMPY*U(N)
    AMVT=AMPY*V(N)
    AMWT=AMPY*W(N)
    DELET=AIX(N)+(U(N)**2+V(N)**2+W(N)**2)/2.
716 DELM=GAMC(J)+AMMY+BMASS(M)-AMPY
    GO TO 715

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713 DELET=AIX(L)+(U(L)**2+V(L)**2+W(L)**2)/2.
715 SIGMU=FLEFT(J)+AMMU-AMUT+BXMOM(M)
SIGMV=YAMC(J)+AMMV-AMVT+BYMOM(M)
DELEK=GAMC(J)*SIGC(J)+AMMY*DELEB-AMPY*DELET+BMASS(M)*BENR(M)
SIGMW=ZMOM(J)+AMMW-AMWT+BZMOM(M)
GO TO 7000

.C
.C      NOW, DO THE SAME FOR THE X DIRECTION
.C      MASS FLUX AT THE RIGHT
.C
.C      *****
7000 IF(AMMP)7003,2702,7002
C
2702 IF(I-IMAX)7017,2703,2703
2703 IF(N2)7016,7017,7016

C
C      FLUX IS POSITIVE
7002 IF(IMAX-I)9914,7004,7005
7017 AMUR=0.
AMVR=0.
AMWR=0.
DELER=0.
GO TO 7100

.C
.C      CHECK CELL TO THE RIGHT
7005 IF(AMX(L+1))9915,7006,7004
.C      FREE SURFACE AT THE RIGHT
7006 IF(AMMP/(DX(I)*DY(J)*DZ(K))-TOZONE)7007,7004,7004

C
C      DENSITY IS TOO SMALL, SET FLUX TO 0.
7007 AMMP=0.
GO TO 7017

C
C      RIGHT FLUX IS NEGATIVE
7003 IF(I-1)9914,7001,7009
7009 IF(AMX(L))9915,7010,7001
7010 IF(-AMMP/(DX(I)*DY(J)*DZ(K))-TOZONE)7011,7001,7001

C
C      FLUX IS TOO SMALL
7011 AMMP=0.
GO TO 7017
7004 DELM=DELM-AMMP
IF(FS)9915,7012,7020

C
C      AT RIGHT BOUNDARY
7020 IF(N2)7150,7014,7150

C
C      RIGHT BOUNDARY IS TRANSMITTIVE
7014 WS=U(L)**2+V(L)**2+W(L)**2
ETH=ETH-AMMP*(AIX(L)+WS/2.)
IF(AMMP/(DX(I)*DY(J)*DZ(K)-Z(77))7012,7012,6640
6640 REZ=1.
7012 AMUR=AMMP*U(L)
AMVR=AMMP*V(L)
AMWR=AMMP*W(L)
DELER=AIX(L)+(U(L)**2+V(L)**2+W(L)**2)/2.
GO TO 7150

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7001 AMUR=AMMP*U(L+1)
    AMVR=AMMP*V(L+1)
    AMWR=AMMP*W(L+1)
    DELER=AIX(L+1)+(U(L+1)**2+V(L+1)**2+W(L+1)**2)/2.
7016 DELM=DELM-AMMP
C
C     SUM UP TOTAL MOMENTA
7150 SIGMU=SIGMU-AMUR
    SIGMV=SIGMV-AMVR
    SIGMW=SIGMW-AMWR
    DELEK=DELEK-AMMP*DELER
C
C     DO THE SAME FOR THE FLUX IN FRONT
C     !!!!!!!!
C
7100 IF(FMASS)7103,2704,9982
2704 IF(K-KMAX)7117,2705,2705
2705 IF(N6)4000,7117,4000
C
C     FLUX IS POSITIVE
9982 IF(KMAX-K)9916,7104,7105
7105 IF(AMX(NN))9917,7106,7104
C
C     FREE SURFACE IN FRONT
7106 IF(FMASS/(DX(I)*DY(J)*DZ(K))-TOZONE)7107,7104,7104
C
C     DENSITY IS TOO SMALL, SET MASS =0.
7107 FMASS=0.
7117 FXMOM=0.
    FYMOM=0.
    FZMOM=0.
    FENR=0.
    GO TO 4000
C
C     FRONT FLUX IS -
7103 IF(K-1)9916,7101,7109
7109 IF(AMX(L))9917,7110,7101
7110 IF(-FMASS/(DX(I)*DY(J)*DZ(K))-TOZONE)7111,7101,7101
C
C     FLUX IS TOO SMALL
7111 FMASS=0.
    GO TO 7117
7104 DELM=DELM-FMASS+AMX(L)
    IF(AREA)9916,7112,7120
7120 IF(N6)8000,7114,8000
C
C     FRONT BOUNDARY IS TRANSMITTIVE
7114 WS=U(L)**2+V(L)**2+W(L)**2
    ETH=ETH-FMASS*(AIX(L)+WS/2.)
    IF(FMASS/(DX(I)*DY(J)*DZ(K))-Z(76))7112,7112,6650
6650 REZ=1.
7112 FXMOM=FMASS*U(L)
    FYMOM=FMASS*V(L)
    FZMOM=FMASS*W(L)
    FENR=AIX(L)+(U(L)**2+V(L)**2+W(L)**2)/2.
    GO TO 8000
7101 FXMOM=FMASS*U(NN)

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FYMOM=FMASS*V(NN)
FZMOM=FMASS*W(NN)
7116 FENR=AIX(NN)+(U(NN)**2+V(NN)**2+W(NN)**2)/2.
4000 DELM=DELM-FMASS+AMX(L)
8000 SIGMU=SIGMU-FXMOM
SIGMV=SIGMV-FY MOM
SIGMW=SIGMW-FZMOM
DELEK=DELEK-FMASS*FENR
C
C      TOTAL MASS AT CYCLE N+1
C      * * * * * * * *
IF(DELM)544,545,540
544 IF(AMX(L)*1.E-6+DELM)9918,545,545
545 DELM=0.
GO TO 550
540 WS=U(L)**2+V(L)**2+W(L)**2
WS=WS/2.
ENK=AMX(L)*(AIX(L)+WS)+DELEK
GO TO 541
C
C      HERE ,WE CALCULATE THE 3 NEW CELL (L) VELOCITIES BY
C      CONSERVING MOMENTUM, WE ALSO CONSERVE THE TOTAL
C      ENERGY, THE TOTAL ENERGY LESS THE KINETIC IS
C      THAN THE NEW SPECIFIC INTERNAL ENERGY.
C
C      NEW X VEL COMPONENT
541 U(L)=(SIGMU+AMX(L)*U(L))/DELM
C
C      NEW Y VEL COMPONENT
546 V(L)=(SIGMV+AMX(L)*V(L))/DELM
C
C      NEW Z VEL COMPONENT
547 W(L)=(SIGMW+AMX(L)*W(L))/DELM
548 WS=U(L)**2+V(L)**2+W(L)**2
C
C      NEW SPECIFIC INTERNAL ENERGY
549 AIX(L)=ENK/DELM-WS/2.
IF(AIX(L)-TMASS)7500,7500,7510
7500 SUM=SUM+AIX(L)*DELM
AIX(L)=0.
7510 IF(ABS(U(L))-XMAX)7501,7501,7502
7501 WS=U(L)**2
SUM=SUM+DELM*WS/2.
U(L)=0.
7502 IF(ABS(V(L))-XMAX)7503,7503,7504
7503 WS=V(L)**2
SUM=SUM+DELM*WS/2.
V(L)=0.
7504 IF(ABS(W(L))-XMAX)7505,7505,7506
7505 WS=W(L)**2
SUM=SUM+DELM*WS/2.
W(L)=0.
7506 IF(AIX(L))4001,550,550
4001 SUM=SUM+AIX(L)*DELM
AIX(L)=0,
550 AMX(L)=DELM
5800 IF(I-I2)5999,5801,5801

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5801 IF(IX2)5999,5802,5999
5802 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5803,5999,5803
5803 IX2=1
      GO TO 5999
5999 IF(K-K2)4999,5804,5804
5804 IF(K22)4999,5805,4999
5805 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5806,4999,5806
5806 K22=1
      GO TO 4999
4999 IF(KZ1)5300,5222,5300
5222 IF(K1-1)5300,5300,5000
5000 IF(K1-K)5300,5001,5001
5001 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5002,5300,5002
5002 KZ1=1
5300 IF(JY1)5600,5304,5600
5304 IF(J1-1)5600,5600,5301
5301 IF(J1-J)5600,5302,5302
5302 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5303,5600,5303
5303 JY1=1
5600 IF(IX1)3342,5604,3342
5604 IF(I1-1)3342,3342,5601
5601 IF(I1-I)3342,5602,5602
5602 IF(ABS(U(L))+ABS(V(L))+ABS(W(L))+AIX(L))5603,3342,5603
5603 IX1=1
3342 CONTINUE
551 IF(AMX(L))9919,553,9980
9980 IF(AMX(L)/(DX(I)*DY(J)*DZ(K))-TOZONE)9981,552,552
9981 AMLOST=AMLOST+AMX(L)
WS=U(L)**2+V(L)**2+W(L)**2
WSR=AMX(L)*(AIX(L)+WS/2.)
SUM=SUM+WSR
ELOST=ELOST+WSR
XMLOST=XMLOST+AMX(L)*U(L)
YMLOST=YMLOST+AMX(L)*V(L)
ZMLOST=ZMLOST+AMX(L)*W(L)
AMX(L)=0.
553 AIX(L)=0.
U(L)=0.
V(L)=0.
W(L)=0.
P(L)=0.

C
C HERE THE FLUX DATA FROM THE RIGHT IS SET TO THE LEFT
552 GAMC(J)=AMMP
FLEFT(J)=AMUR
YAMC(J)=AMVR
ZMOM(J)=AMWR
SIGC(J)=DELER

C
C HERE THE FLUX DATA FROM THE TOP IS SET TO THE BOTTOM
C
554 AMMY=AMPY
AMMU=AMUT
AMMV=AMVT
AMMW=AMWT
DELEB=DELET
C

```

C HERE THE FLUX DATA FROM IN FRONT IS SET TO THE BACK

555 BMASS(M)=FMASS  
 BYMOM(M)=FYMOM  
 BXMOM(M)=FXMOM  
 BZMOM(M)=FZMOM  
 BENR(M)= FENR  
 CONTINUE  
 L=L+IMAX

C \$\$\$\$\$\$\$\$\$\$\$\$\$ END OF J LOOP \$

3 CONTINUE

5700 LJ=L-IMAX  
 IF(JY2)4,5701,4  
 5701 IF(ABS(U(LJ))+ABS(V(LJ))+ABS(W(LJ))+AIX(LJ))5702,4,5702  
 5702 JY2=1

4 CONTINUE

556 CONTINUE

C \$\$\$\$\$\$\$\$\$\$\$\$\$ END OF I LOOP \$\$\$\$\$\$\$\$\$\$\$\$\$

2 CONTINUE

557 CONTINUE

C \$ END OF K LOOP \$

1 CONTINUE

8001 ETH=ETH-SUM  
 ENEG=ENEQ-SUM  
 I1=I1-IX1

5900 IF(I1)5901,5901,5902  
 5901 I1=1

5902 I2=I2+IX2

5903 IF(I2-IMAX)5905,5905,5904  
 5904 I2=IMAX  
 5905 J1=J1-JY1  
 IF(J1)5906,5906,5907

5906 J1=1

5907 J2=J2+JY2  
 IF(J2-JMAX)5909,5909,5908

5908 J2=JMAX  
 5909 K1=K1-KZ1  
 IF(K1)5910,5910,5911

5910 K1=1

5911 K2=K2+KZ2  
 IF(K2-KMAX)5913,5913,5912

5912 K2=KMAX  
 5913 IF(REZ)9950,9950,9951

9951 CALL REZONE

9950 RETURN

9903 NK=600  
 GO TO 9999

9904 NK=601  
 GO TO 9999

8999 NK=9  
 GO TO 9999

9900 NK=15  
 GO TO 9999

9901 NK=506  
 GO TO 9999

9902 NK=23  
 GO TO 9999

9910 NK=1103

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      GO TO 9999
9911 NK=106
      GO TO 9999
* 9912 NK=317
      GO TO 9999
9913 NK=702
      GO TO 9999
9914 NK=705
      GO TO 9999
9915 NK=7005
      GO TO 9999
9916 NK=9982
      GO TO 9999
9917 NK=7105
      GO TO 9999
9918 NK=544
      GO TO 9999
9919 NK=551
9999 NR=4
      WRITE(6,8500)I,J,K,L,M,N,NN,NK,NR
      WRITE(6,8501)GAMC(J),FLEFT(J),YAMC(J),ZMOM(J),SIGC(J)
      WRITE(6,8501)AMMP,AMUR,AMVR,AMWR,DELER
      WRITE(6,8501)AMMY,AMMU,AMMV,AMMW,DELEB
      WRITE(6,8501)AMPY,AMUT,AMVT,AMWT,DELET
      WRITE(6,8501)BMASS(M),BXMOM(M),BYMOM(M),BZMOM(M),BENR(M)
      WRITE(6,8501)FMASS,FXMOM,FYMOM,FZMOM,FENR
      WRITE(6,8501)AMX(L),AIX(L),U(L),V(L),W(L),P(L),CYCLE
      WRITE(6,8501)AMX(L+1),U(L+1),V(L+1),W(L+1)
      WRITE(6,8501)AMX(NN),U(NN),V(NN),W(NN)
      WRITE(6,8501)AMX(L-1),U(L-1),V(L-1),W(L-1)
      WRITE(6,8501)AMX(N),U(N),V(N),W(N)
      LL=L-IMAX
      LBJ=L-IXMAX
      WRITE(6,8501)AMX(LL),U(LL),V(LL),W(LL)
      WRITE(6,8501)AMX(LBJ),U(LBJ),V(LBJ),W(LBJ)
8501 FORMAT(1P8E12.5)
8500 FORMAT(9I6)
      CALL UNCLE
      CALL DUMP
      END
@I FOR REZONE/S,REZONE/S,REZONE/SS
      SUBROUTINE REZONE
C      CHANGE ALL CELL DIMENSIONS BY A FACTOR OF 2.
C      NOTE, 8 CELLS BECOME ONE IN THE NEW GRID
C      CALCULATE NEW INDICES
      KKMAX=KMAX/2
      IIMAX=IMAX/2
      JJMAX=JMAX/2
C      SET UP DO LOOP FOR NEW STORAGE
      KN1=-IXMAX
      DO 21 KKK=1,KMAX,2
      LL=(KKK-1)*IXMAX
      KN1=KN1+IXMAX
      KN=KN1
      DO 21 II=1,IMAX,2
      I=LL+II
      KN=KN+1

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```

KK=KN
DO 21 JJ=1,JMAX,2
K=I+IXMAX
J=I+IMAX
L=K+IMAX
WSA=AMX(I)+AMX(I+1)+AMX(J)+AMX(J+1) +
1AMX(K)+AMX(K+1)+AMX(L)+AMX(L+1)
IF(WSA)6,6,3
C CALCULATE TWICE THE KINETIC ENERGY
3 WSB=AMX(I)*(U(I)**2+V(I)**2+W(I)**2) +
1AMX(I+1)*(U(I+1)**2+V(I+1)**2+W(I+1)**2) +
2AMX(J)*(U(J)**2+V(J)**2+W(J)**2)+AMX(J+1)*(U(J+1)**2
3+V(J+1)**2+W(J+1)**2)
WSB=WSB+AMX(K)*(U(K)**2+V(K)**2+W(K)**2)+AMX(K+1)*
1(U(K+1)**2+V(K+1)**2+W(K+1)**2)+AMX(L)*(U(L)**2+V(L)**2
2+W(L)**2)+AMX(L+1)*(U(L+1)**2+V(L+1)**2+W(L+1)**2)
C CALCULATE THE NEW VELOCITIES
4 U(KK)=(AMX(I)*U(I)+AMX(I+1)*U(I+1)+AMX(J)*U(J)+AMX(J+1)*U(J+1) +
1AMX(K)*U(K)+AMX(K+1)*U(K+1)+AMX(L)*U(L)+AMX(L+1)*U(L+1))/WSA
V(KK)=(AMX(I)*V(I)+AMX(I+1)*V(I+1)+AMX(J)*V(J)+AMX(J+1)*V(J+1) +
1AMX(K)*V(K)+AMX(K+1)*V(K+1)+AMX(L)*V(L)+AMX(L+1)*V(L+1))/WSA
W(KK)=(AMX(I)*W(I)+AMX(I+1)*W(I+1)+AMX(J)*W(J)+AMX(J+1)*W(J+1) +
1AMX(K)*W(K)+AMX(K+1)*W(K+1)+AMX(L)*W(L)+AMX(L+1)*W(L+1))/WSA
C CALCULATE THE TOTAL INTERNAL ENERGY
WSC=AMX(I)*AIX(I)+AMX(I+1)*AIX(I+1)+AMX(J)*AIX(J) +
1AMX(J+1)*AIX(J+1)+AMX(L)*AIX(L)+AMX(L+1)*AIX(L+1)+AMX(K)*
2AIX(K)+AMX(K+1)*AIX(K+1)
P(KK)=0.
C SET THE NEW MASSES
AMX(KK)=WSA
C CALCULATE THE NEW KINETIC ENERGY (ACTUALLY
TWICE)
WS=U(KK)**2+V(KK)**2+W(KK)**2
E=WSC+WSB/2.
C THE NEW SPECIFIC INTERNAL ENERGY IS THE
TOTAL LESS THE KINETIC
AIX(KK)=E/WSA-.5*WS
AMX(J)=0.
AMX(J+1)=0.
AMX(K)=0.
AMX(K+1)=0.
AMX(L)=0.
AMX(L+1)=0.
AMX(I+1)=0.
U(J)=0.
U(J+1)=0.
U(K)=0.
U(K+1)=0.
U(L)=0.
U(L+1)=0.
U(I+1)=0.
V(J)=0.
V(J+1)=0.
V(K)=0.
V(K+1)=0.
V(L)=0.
V(L+1)=0.

```

```

V(I+1)=0.
W(J)=0.
W(J+1)=0.
W(K)=0.
W(K+1)=0.
W(L)=0.
W(L+1)=0.
W(I+1)=0.
AIX(I+1)=0.
AIX(J)=0.
AIX(J+1)=0.
AIX(K)=0.
AIX(K+1)=0.
AIX(L)=0.
AIX(L+1)=0.
IF(II-1)380,380,390
380 IF(JJ-1)381,381,390
381 IF(KK-1)7,7,390
390 AMX(1)=0.
U(I)=0.
V(I)=0.
W(I)=0
P(I)=0.
AIX(I)=0.
GO TO 7
C CAME HERE BECAUSE OF ZERO MASS
6 AMX(KK)=0.
U(KK)=0.
V(KK)=0.
W(KK)=0.
P(KK)=0.
AIX(KK)=0.
P(KK)=0.
7 KK=KK+IMAX
21 I=I+2*IMAX
C CHANGE ALL CELL DIMENSIONS
WS=0.
DO 10 I=1,IIMAX
DX(I)=2.0*DX(I)
X(I)=WS+DX(I)
WS=X(I)
10 CONTINUE
II=IIMAX+1
WS=X(IIMAX)
DO 11 I=II,IMAX
DX(I)=DX(IIMAX)
X(I)=WS+DX(I)
WS=X(I)
11 CONTINUE
WS=0.
DO 13 J=1,JJMAX
DY(J)=2.0*DY(J)
Y(J)=WS+DY(J)
WS=Y(J)
13 CONTINUE
JJ=JJMAX+1
WS=Y(JJMAX)

```

```

DO 14 J=JJ,JMAX
DY(J)=DY(JJMAX)
Y(J)=WS+DY(J)
WS=Y(J)
14 CONTINUE
WS=0.
DO 16 K=1,KKMAX
DZ(K)=2.0*DZ(K)
ZCOR(K)=WS+DZ(K)
WS=ZCOR(K)
16 CONTINUE
KK=KKMAX+1
WS=ZCOR(KKMAX)
DO 17 K=KK,KMAX
DZ(K)=DZ(KKMAX)
ZCOR(K)=WS+DZ(K)
WS=ZCOR(K)
17 CONTINUE
KK=KKMAX+1
DO 30 K=KK,KMAX
LL=(K-1)*IXMAX
DO 30 I=1,IMAX
L=LL+I
DO 30 J=1,JMAX
AMX(L)=0.
U(L)=0.
V(L)=0.
W(L)=0.
AIX(L)=0.
P(L)=0.
30 L=L+IMAX
DO 100 K=1,KKMAX
LL=(K-1)*IXMAX
DO 100 I=1,IIMAX
L=LL+IIMAX+1-I
M=LL+IMAX-I4+1-I
DO 100 J=1,JJMAX
AMX(M)=AMX(L)
U(M)=U(L)
V(M)=V(L)
W(M)=W(L)
AIX(M)=AIX(L)
P(M)=P(L)
AMX(L)=0.
U(L)=0.
V(L)=0.
W(L)=0.
AIX(L)=0.
P(L)=0.
M=M+IMAX
L=L+IMAX
100 CONTINUE
I3=I3/?
C NOW BEGIN ADDING ON MASS IN FRONT, BOTH
C SIDES AND ABOVE.
II=IIMAX-I4
JJ=I3+1

```

C NOTE, I4=NO. OF ZONES TO THE RIGHT TO ADD.  
 C I3= INITIAL INTERFACE BETWEEN PROJECTILE AND TARGET.

```

DO 200 K=1,KMAX
LL=(K-1)*IXMAX
DO 200 I=1,II
L=LL+I+I3*IMAX
DO 200 J=JJ,JMAX
AMX(L)=DX(I)*DY(J)*DZ(K)*RHONOT
200 L=L+IMAX
II=IMAX-I4
IL=IIMAX-I4+1
JL=JJMAX+1
DO 300 K=1,KMAX
LL=(K-1)*IXMAX
DO 300 I=IL,II
L=LL+I+JJMAX*IMAX
DO 300 J=JL,JMAX
AMX(L)=DX(I)*DY(J)*DZ(K)*RHONOT
300 L=L+IMAX
II=IMAX-I4+1
JJ=I3+1
DO 400 K=1,KMAX
LL=(K-1)*IXMAX
DO 400 I=II,IMAX
L=LL+I+I3*IMAX
DO 400 J=JJ,JMAX
AMX(L)=DX(I)*DY(J)*DZ(K)*RHONOT
400 L=L+IMAX
II=IMAX-I4
IL=IIMAX-I4+1
JJ=I3+1
IF(JJ-JJMAX)700,700,800
700 KL=KKMAX+1
DO 500 K=KL,KMAX
LL=(K-1)*IXMAX
DO 500 I=IL,II
L=LL+I+I3*IMAX
DO 500 J=JJ,JMAX
AMX(L)=DX(I)*DY(J)*DZ(K)*RHONOT
500 L=L+IMAX
800 WS=T+DTNA
C      RESET ACTIVE GRID COUNTERS
I1=I4-1
I2=IMAX-I4+2
J1=J1
J2=JJMAX+2
K1=K1
K2=KKMAX+2
NK=NC+1
WRITE(6,9000)WS,NK
9000 FORMAT(1H ////22H PROBLEM REZONED AT T=1PE12.6,6X,5HCYCLEI4////)
WRITE(6,8000)IMAX,(X(I),I=1,IMAX)
WRITE(6,8003)IMAX,(DX(I),I=1,IMAX)
WRITE(6,8001)JMAX,(Y(J),J=1,JMAX)
WRITE(6,8004)JMAX,(DY(J),J=1,JMAX)
WRITE(6,8002)KMAX,(ZCOR(K),K=1,KMAX)
WRITE(6,8005)KMAX,(DZ(K),K=1,KMAX)

```

```

      WRITE(6,8006) IMAX,JMAX,KMAX,IXMAX,KMAXA
      RETURN
8000 FORMAT(1H /10H X(I) I=1,I2/(5F16.6))
8001 FORMAT(1H /10H Y(J) J=1,I2/(5F16.6))
8002 FORMAT(1H /13H ZCOR(K) K=1,I2/(5F16.6))
8003 FORMAT(1H /11H DX(I) I=1,I2/(5F16.6))
8004 FORMAT(1H /11H DY(J) J=1,I2/(5F16.6))
8005 FORMAT(1H /11H DZ(K) K=1,I2/(5F16.6))
8006 FORMAT(7I8)
      END
      QI FOR   ES/S,ES/S,ES/SS
      SUBROUTINE ES

```

ES 0010  
PH2 0690

C C \*\*\*\* \* 3DOIL \*\*\*\* \*

C C \$\$\$\$ FOR COMPLETE DETAILS, SEE GA-3216,  
 METALLIC EQUATIONS OF STATE FOR HYPERVELOCITY IMPACT  
 BY JAMES TILLOTSON  
 IF THE MATERIAL IS COMPRESSED , USE THE CONDENSED  
 FORM OF THE EQUATION OF STATE.  
 IF THE MATERIAL IS RAREFIED AND IF THE SPECIFIC  
 INTERNAL ENERGY IS GREATER THAN E SUB S , USE  
 THE RARFIED FORM ... BUT-----  
 IF RAREFIED AND E IS LESS THAN E SUB S,  
 USE THE CONDENSED FORM.  
 \$\$\$\$ NOTE, NO NEGATIVE(TENSION) PRESSURES ALLOWED

ES 0980

```

10 RHOW=AMX(L)/(DX(I)*DY(J)*DZ(K))
ETA=RHOW/Z(33)
VOW=1.0/ETA
11 P1=AIX(L)*RHOW*Z(34)
12 P2=AIX(L)
13 P3=Z(35)*ETA*ETA
14 P4=Z(36)/(P2/P3+1.0)*AIX(L)*RHOW
15 P5=Z(37)*(ETA-1.)
16 IF(ETA-1.0)50,100,100
50 IF(VOW-Z(38))55,55,75
55 IF(AIX(L)-Z(40))100,100,75
75 P7=Z(41)*(VOW-1.)
IF(P7-88.0)4002,4002,4003
4003 P7=88.0
4002 CONTINUE
P8=EXP(P7)
P9=1.0/P8
P10=Z(42)*((VOW-1.)**2)
IF(P10-88.0)4000,4000,4001
4001 P10=88.0
4000 CONTINUE
P11=EXP(P10)
P12=1.0/P11
P(L)=P1+(P4+P5*P9)*P12
GO TO 119
100 P6=Z(44)*((ETA-1.)**2)
P(L)=P1+P4+P5+P6

```

ES 1010

ES 1070

ES 1110

ES 1120

ES 1130

ES 1140

ES 1150

ES 1170

ES 1180

ES 1190

ES 1200

ES 1210

```

119 IF(P(L))999,999,200
200 WSGX=.5
    GO TO 500
ES   1260
999 P(L)=0.
WSGX=.5+Z(43)
GO TO 500
ES   1270
500 RETURN
END
ES   1300

D1 FOR EDIT/S,EDUIT/S,EDIT/SS
SUBROUTINE EDIT
C
C **** 3DOIL ****
C HERE WE WILL DECIDE WHETHER TO HAVE A SHORT PRINT, LONG PRINT,
C DUMP ON THE BINARY TAPE OR STOP THE PROBLEM
C
104 CALL SLITET(3,K000FX)
    GO TO(106,108),K000FX
105 CALL SLITE (3)
    GO TO 126
106 CALL SLITE (3)
    GO TO 126
108 IF(CYCLE-CSTOP)110,122,122
110 IF(REZ)9901,112,124
112 IF(AMOD(CYCLE,DUMPT7))114,124,114
114 IF(AMOD(CYCLE,PRINTL))120,126,120
120 IF(AMOD(CYCLE,PRINTS))140,128,140
122 CALL SLITE (1)
124 GO TO 1
126 CALL SLITE (4)
128 GO TO 6000
130 GO TO 1000
132 CALL SLITET(4,K000FX)
    GO TO(134,136),K000FX
134 GO TO 5000
EDIT1040
EDIT1050
EDIT1060
EDIT1070
EDIT1080
EDIT1100
EDIT1110
EDIT1120
EDIT1130
EDIT1140
EDIT1150
EDIT1160
EDIT1170
EDIT1180
EDIT1190
EDIT1200
EDIT1210
EDIT1220
EDIT1230

C
C CALCULATE THE ENERGY CHECK = (ETH-E)/ETH AT CYCLE N+1 -
C (ETH-E)/ETH AT CYCLE M ALL DIVIDED THRU BY THE NUMBER OF CYCLE
C BETWEEN ENERGY CHECKS =(CYCLE(N+1) -CYCLE M)
C
136 IF(ABS(ECK)-DMIN)140,140,9905
140 CALL SLITET(1,K000FX)
    GO TO(142,144),K000FX
142 REWIND(N7)
    CALL SLITE (1)
144 GO TO 10000
1 IF(DUMPT7)30,3,3
EDIT1240
EDIT1250
EDIT1260
EDIT1280
EDIT1290
EDIT1330

C
C **** DUMP ALL THE CELL-CENTERED QUANTITIES ,THE X,S AND Y,S
C AND ZCOR,S AND THE ENTIRE Z BLOCK
C
3 BACKSPACE N7
WS=555.0
WRITE(N7)WS,CYCLE,N3
EDIT1360
WRITE(N7)(Z(L),L=1,M2)
WRITE(N7)(U(K),V(K),W(K),AMX(K),AIX(K),K=1,KMAXA)
WRITE(N7)(X(I),I=1,IMAX)
WRITE(N7)(Y(J),J=1,JMAX)
WRITE(N7)(ZCOR(K),K=1,KMAX)
WS=666.0
EDIT1480

```

```

      WRITE(N7)WS,WS,WS
      WRITE(6,8120)NC
. 30 GO TO 126 EDIT1510
. 6000 CONTINUE
.     DO 6012 I=1,18
. 6012 PR(I)=0.
. C
. C      CALCULATE THE TOTAL INTERNAL AND KINETIC ENERGY ,AND THE
. C      TOTAL MASS.
. C
.     DO 6028 K=1,KMAXA
6017 PR(1)=0.0 EDIT1760
PR(2)=0.0 EDIT1770
PR(4)=0.
6019 IF(AMX(K))9917,6028,6020 EDIT1790
6020 WSB=(U(K)**2+V(K)**2+W(K)**2)*.5
      PR(5)=PR(5)+AIX(K)*AMX(K)
      PR(6)=PR(6)+WSB*AMX(K)
      PR(8)=PR(8)+AMX(K)
6028 CONTINUE
      PR(3)=PR(1)+PR(2)
      PR(7)=PR(5)+PR(6)
      XNRG=PR(7)
      PR(9)=PR(1)+PR(5)
      PR(10)=PR(2)+PR(6)
      PR(11)=PR(3)+PR(7)
      PR(12)=PR(4)+PR(8)
      WSA=(ETH-PR(11))/ETH
      IF(CYCLE)9931,9931,9932
9931 NPC=1
9932 PR(18)=(WSA-DNN)/FLOAT(NPC)
      ECK=PR(18) EDIT2040
      DNN=WSA
      NPC=0
      SUM=0.0 EDIT2050
C      RADEB= TOTAL POSITIVE Z MOM.
C      RADER = TOTAL POSITIVE X MOM.
C      RADET= TOTAL POSITIVE Y MOM.
      SUMB=0.
      SUMR=0.
      SUMT=0.
      DO 810 K=1,KMAXA
      IF(AMX(K))810,810,802
802 IF(V(K))804,804,803
803 SUMT=SUMT+AMX(K)*V(K)
804 IF(U(K))805,805,806
806 SUMR=SUMR+AMX(K)*U(K)
805 IF(W(K))810,810,808
808 SUMB=SUMB+AMX(K)*W(K)
810 CONTINUE
      RADEB=SUMB
      RADER=SUMR
      RADET=SUMT
      WRITE(6,8116)PROB,NC,T,DTNA,TRAD,DTRAD,NR,N1,N2,N3,N4
      WRITE(6,8117)(PR(I),I=1,8)
      WRITE(6,8118)(PR(I),I=9,12)
      WRITE(6,8119)RADEB,RADER,RADET,UVMAX,ETH,ECK

```

```

      WRITE(6,9040)N10,N11,N9,I1,I2,J1,J2,K1,K2
      WRITE(6,9042)AMLOST,ELOST,XMLOST,YMLOST,ZMLOST,ENEQ
9042 FORMAT(1P7E14.7)
6090 GO TO 130
1000 GO TO 1030
1030 WRITE(6,8116)PROB,NC,T,DTNA,TRAD,DTRAD,NR,N1,N2,N3,N4
      JMAX=JMAX
      WRITE(6,8307)X1,X2,XMAX,Y1,Y2,Y(JMAX)
      GO TO 132
5000 WRITE(6,8116)PROB,NC,T,DTNA,TRAD,DTRAD,NR,N1,N2,N3,N4
C   NOTICE THE LIMITS OF THE DO LOOPS
DO 1126 KK=K1,K2
C
C   HERE WE  PREPARE FOR THE LONG PRINT
C
      WRITE(6,9041)KK,ZCOR(KK),DZ(KK)
5004 DO 5050 I=I1,I2
      CALL SLITE (4)
      J=J2+1
      K=J2*IMAX+I+(KK-1)*IXMAX
      DO 5046 L=J1,J2
      J=J-1
      K=K-IMAX
5012 IF(AMX(K))9917,5046,5014
5014 CALL SLITET(4,K000FX)
      GO TO(5016,5019),K000FX
5016 WRITE(6,8135)I,X(I),DX(I)
5019 WSC=AMX(K)/(DX(I)*DY(J)*DZ(KK))
      WSC=P(K)*1.E+4
      EDIT3050
C
C   J=ROW NUMBER
C
C   U= X COMPONENT OF VELOCITY  CM./SH.
C
C   V= Y COMPONENT OF VELOCITY  CM./SH.
C
C   WSC = PRESSURE IN MEGABARS
C
C   AMX = MASS IN GRAMS
C
C   WS= DENSITY IN GRAMS/CM CUBED
C
C   W= Z COMPONENT OF VELOCITY  CM./SH.
C
C   AIX= SPECIFIC INTERNAL ENERGY IN  JERKS/GM. (1 JERK= 10(16)ERGS)
C
C   Y= TOP COORDINATE OF THIS ROW  IN CM.
C
C
5018 WRITE(6,8108)J,U(K),V(K),WSC,AMX(K),WS,AIX(K),W(K),Y(J)
5046 CONTINUE
5050 CONTINUE
1126 CONTINUE
      GO TO 136
9901 NK=110
      GO TO 9999
9905 NK=136
      EDIT3080
      EDIT3090
      EDIT3100
      EDIT3150
      EDIT3160
      EDIT3180

```

GO TO 9999 EDIT3190  
 9917 NK=6015 EDIT3210  
 GO TO 9999 EDIT3220  
 9999 NR=6 EDIT3320  
 WRITE(6,8002)I,J,K,KP,I1,I2,NK,NR  
 8002 FORMAT(8I5)  
 CALL UNCLE  
 CALL DUMP  
 10000 RETURN EDIT3340  
 C EDIT3350  
 C FORMATS EDIT3360  
 8108 FORMAT(I3,1X,1P8E12.5)  
 81160FORMAT(8H1PROBLEM&X,5HCYCLE9X,4HTIME13X,2HDT13X,4HTRAD11X,5HDTRAD1EDIT3380  
 12X,2HNR6X,2HN14X,2HN24X,2HN34X,2HN4/(F7.1,I11,2X,1P4E16.7,I10,2X,4EDIT3390  
 216)) EDIT3400  
 81170FORMAT(1H0//17X2HA.I16X,2HAK14X,5HAI+AK15X,2HAM/4H DOT3X,1P4E18.7/3EDIT3410  
 1H X4X,4E18.7) EDIT3420  
 81180FORMAT(12X,13H-----5X,13H-----5X,13H-----5EDIT3430  
 1X,13H-----/7H TOTALS1P4E18.7) EDIT3440  
 81190FORMAT(2H0 //16X,5HRADEB13X,5HRADER13X,5HRADET12X,7HMAX VEL13X,3HEDIT3450  
 1HE12X,9HREL ERROR/7X,1P6E18.7////) EDIT3460  
 8120 FORMAT(1H0//21H TAPE 7 DUMP ON CYCLEI5////) EDIT3470  
 81350FORMAT(1H //4H I =I3.6X,6HX(I) =F12.3,6X,7HDX(I) =F12.3//3H J8X,EDIT3520  
 11HX10X,1HY10X,3HF/A9X,3HAMX9X,3HRH08X,3HAI9X,4H W 8X,2H Y/)  
 8307 FORMAT(5H X1 =1PE12.5,3X,4HX2 =E12.5,3X,6HXMAX =E12.5,6X,4HY1 =E12  
 1.5,3X,4HY2 =E12.5,3X,6HYMAX =E12.5)  
 9040 FORMAT(1H / 9I6)  
 9041 FORMAT(1H //4H K =I3.6X,9HZCOR(K) =F12.3,6X,7HDZ(K) =F12.3)  
 END EDIT3630

## 5. INPUT AND DEFINITIONS OF THE VARIABLES

### 5.1 Normal Input for the TRIOIL Code

An asterisk (\*) implies that the data on the card is to be converted to fixed point data (requires a 2 punch in Column 1). All data loaded via the card routine is read by a floating point format. A double asterisk (\*\*) signifies that this is the last data card in this set, and requires a one in Column 1.

#### First Set

The number of BCD (header cards) that will be read in (Columns 1-3, format I3).

N BCD cards, alphabetic and or numeric in Columns 2-72.

#### Second Set

<u>Location</u>	<u>Symbol</u>	<u>Description</u>
*103	N7	Binary tape number (data tape)
**36271	PK(1)	Problem number
36272	PK(2)	The cycle number to start the calculation
36273	PK(3)	If < 0. code assumes that this is a re-start or that the CLAM code has generated the initial data. (NOTE: At this time of writing the report, a three-dimensional version of the CLAM code is not available.) If PK(3) $\geq$ , the code will call subroutine set-up.

#### Third Set

<u>Location</u>	<u>Symbol</u>	<u>Description</u>
1	PROB	Problem number, identical to the value in PK(1)
3	DT	The time step $\Delta t^n$ in shakes. (NOTE: 1 shake = $10^{-8}$ seconds)
4	PRINTS	Short print frequency in cycles
5	PRINTL	Long print frequency in cycles
6	DUMPT7	Binary tape dump frequency in cycles

<u>Location</u>	<u>Symbol</u>	<u>Description</u>
7	CSTOP	Cycle at which the problem will stop
13	FFA	Upper limit for stability and to calculate $\Delta t$ if CABLН = 0.
14	FFB	Lower limit for stability and to calculate $\Delta t$ if CABLН = 0.
20	DMIN	If ECK (Z(24)) > DMIN, problem will stop because of poor energy conservation
23	TOZONE	$\sim 10^{-4} \rho_0$ = minimum density for mass transport at a free surface within the grid
25	SBOUND	= 1.0, fraction of $\Delta$ in. the weighted velocity term in the calculation of the mass flux
26	CABLН	<p>If &lt; 0. the code will control <math>\Delta t</math> at PCSTAB (an input number) of stability</p> <p>If = 0. the code will control <math>\Delta t</math>, decreasing <math>\Delta t</math> if <math> u \text{ or } c  \Delta t / \Delta x</math> or <math> v \text{ or } c  \Delta t / \Delta y</math> or <math> w \text{ or } c  \Delta t / \Delta z</math> exceed FFA (an input number) and increasing <math>\Delta t</math> if the above terms are less than FFB (an input number)</p> <p>If &gt; 0., the <math>\Delta t</math> that is loaded at <math>t = 0.</math>, will remain constant, regardless of any stability considerations</p>
NOTE: These 2 options require that you load $\Delta t$ (location 3) at time $t = 0.$ .		
29	WSGD	= GAMMA for the dot material
30	WSGX	= GAMMA for the x material
45	DTCHK	$\sim 10^{-4} \rho_0$ , any cell that has a density less than this value, will be bypassed for stability checks
46	PCSTAB	$\sim .25$ , fraction of stability as determined by the Courant condition or particle velocity
58	Z(58)	Initial x velocity component of the projectile in cm/shake
59	Z(59)	Initial z velocity component of the projectile in cm/shake
66	RHONOT	Initial density ( $\rho_0$ ) of all material (since this is a one-material code)
67	VELOC	Initial y velocity component of the projectile in cm/shake
68	BUG	$\sim .05$ , epsilon for determining whether special features will be used to empty the bottom cells in the projectile

<u>Location</u>	<u>Symbol</u>	<u>Description</u>
69	Z(69)	= percent of (PCSTAB) at time t = 0. used for problems in which most of the energy at t = 0. is in the form of internal energy
70	Z(70)	Factor to increase Z(69) every cycle up until Z(69) is equal to 1.0
*86	iMAX	Maximum number of zones in the x direction
*87	jMAX	Maximum number of zones in the y direction
*88	kMAX	Maximum number of zones in the z direction
75	Z(75)	Density of material leaving the left boundary of grid in order to trigger rezone
76	Z(76)	Similar term for the front boundary of grid
77	Z(77)	Similar term for the right boundary of grid
78	Z(78)	Similar term for the back boundary of grid
79	Z(79)	Similar term for the top boundary of grid
80	Z(80)	Similar term for the bottom boundary of grid
*95	i3	The original (j) value of projectile-target interface
*96	i4	The number of zones to add on to the right side of grid after rezone. i4 + the number to add on the left is = to iMAX/2
*97	N1	If = 0. the left side of grid is transmittive, otherwise reflective
*98	N2	Similar term for the right side of grid
*99	N3	Similar term for the top side of grid
*100	N4	Similar term for the bottom side of grid
*101	N5	Similar term for the back side of grid
*102	N6	Similar term for the front side of grid
*103	N7	Binary tape number
36151	Dx	= $\Delta x$ to be used for all (i)
36181	Dy	= $\Delta y$ to be used for all (j)
36211	Dz	= $\Delta z$ to be used for all (k)
50	S1	= interface (j) value + 1, between the projectile and the target
51	S2	= back (k) boundary + 1 of the projectile
52	S3	= front (k) boundary of the projectile
53	S4	= the left (i) boundary + 1, of the projectile

<u>Location</u>	<u>Symbol</u>	<u>Description</u>
54	S5	= the right (i) boundary of the projectile
55	S6	= the bottom (j) boundary + 1, of the projectile
56	S7	= the top boundary of the projectile (j)
*93	i1 }	= minimum and maximum values of the active grid in the x direction
*94	i2 }	
*108	k1 }	= minimum and maximum values of the active grid in the z direction
*109	k2 }	
*110	j1 }	= minimum and maximum values of the active grid in the y direction
*111	j2 }	
47	Z(47)	= $c_o$
48	Z(48)	= $A_1$
49	Z(49)	= $A_2$
		Where P is in megabars and $c_o$ and $A_1$ are in units of $10^7$ cm/sec, converted to cm/shake in CDT routine
33	Z(33)	= $\rho_o$
34	Z(34)	= a
35	Z(35)	= $E_o$
36	Z(36)	= b
37	Z(37)	= A
38	Z(38)	= $V_s$
39	Z(39)	Blank
40	Z(40)	= $E_s$
41	Z(41)	= $\alpha$
42	Z(42)	= $\beta$
43	Z(43)	Blank
44	Z(44)	= B
16	xMAX	= epsilonics on the velocity, if $ u $ or $ v $ or $ w  < \epsilon$ , it is set to 0. and the books are kept
15	TMASS	= epsilonics on the specific internal energy. If $I < \epsilon$ , it is set to 0. and the books are kept
**22	REZFCT	No meaning in this code

Last Set

<u>Location</u>	<u>Symbol</u>	<u>Description</u>
**22	REZFCT	No meaning

5.2 List of Common for TRIOIL

<u>Symbol</u>	<u>Location</u>	<u>No. of Words</u>	<u>Units</u>	<u>Description</u>
AIX	151	6000	jerks/g	Specific internal energy for cell (L)
AMX	6151	6000	g	Mass in cell (L)
U	12151	6000	cm/shake	X component of velocity in cell (L)
V	18151	6000	cm/shake	Y component of velocity in cell (L)
W	24151	6000	cm/shake	Z component of velocity in cell (L)
P	30151	6000	jerks/cm <sup>3</sup>	Material pressure in cell (L)
DX	36151	30	cm	DX(i) = X(i) - X(i-1)
DY	36181	30	cm	DY(j) = Y(j) - Y(j-1)
DZ	36211	30	cm	DZ(k) = ZCOR(k) - ZCOR(k-1)
UL	36241	30	cm/shake	Velocity at the left of cell in PH1
FLEFT	36241	30	g cm/shake	X momenta of mass crossing left side of cell (PH2)
PL	36271	30	jerks/cm <sup>3</sup>	Temporary pressure array at the left interface in PH1
PK	36271	30	none	Not used (except input)
YAMC	36271	30	g cm/shake	Y momenta of mass crossing left side of cell PH2
X	36301	30	cm	X(i) = right dimension of cell (L)
Y	36331	30	cm	Y(j) = top dimension of cell (L)
ZCOR	36361	30	cm	ZCOR(k) = Z or front dimension of cell (L)
PR	36391	50	many	Used for editing in the EDIT routine
SIGC	36441	30	jerks/g	= specific energy of the mass crossing left side of cell (PH2)
GAMC	36471	30	g	Mass crossing left side of cell in (PH2)

<u>Symbol</u>	<u>Location</u>	<u>No. of Words</u>	<u>Units</u>	<u>Description</u>
ZMOM	36501	30	g cm/shake	Z momenta of mass crossing left side of cell (PH2)
BXMOM	36531	700	g cm/shake	X momenta of mass crossing back side of cell (PH2)
UBIND	36531	700	cm/shake	Velocity at back interface of cell in PH1
PBIND	37231	700	jerks/cm <sup>3</sup>	Pressure array at back interface of cell in (PH1)
BMASS	37231	700	g	Mass crossing back interface of cell in PH2
BYMOM	37931	700	g cm/shake	Y momenta of mass crossing back surface in PH2
BZMOM	38631	700	g cm/shake	Z momenta of mass crossing back surface in PH2
BENR	39331	700	jerks/g	Specific energy of this mass crossing the back surface in PH2
AREA	40031	1	none	Flag in PH2 for the Z direction
BIG	40032	1	none	Not used
BOUNCE	40033	1	none	Not used
PABOVE	40034	1	jerks/cm <sup>3</sup>	Pressure at the top of cell (L) in PH1
PBLO	40035	1	jerks/cm <sup>3</sup>	Pressure at the bottom of cell (L) in PH1
P1DTS	40036	1	none	Not used
PRR	40037	1	jerks/cm <sup>3</sup>	Pressure at the right of cell (L) in PH1
RHO	40038	1	none	Not used
SIG	40039	1	cm	Minimum Δx, Δy or Δz for a cell in CDT routine
UVMAX	40040	1	shake <sup>-1</sup>	(Maximum velocity)/minimum (Δx or Δy or Δz)
VABOVE	40041	1	cm/shake	Velocity at the top of cell (L) in PH1
VBLO	40042	1	cm/shake	Velocity at the bottom of cell (L) in PH1
VEL	40043	1	none	Maximum (α-1) in the CDT routine, a flag for subcycling in PH1 and a flag for the y direction in PH2

<u>Symbol</u>	<u>Location</u>	<u>No. of Words</u>	<u>Units</u>	<u>Description</u>
WPS	40044	1	none	Working storage
WS	40045	1	none	Working storage
WSA	40046	1	none	Working storage
WSB	40047	1	none	Working storage
WSC	40048	1	none	Working storage
i	40049	1	none	Temporary indices
ii	40050	1	none	Temporary indices
iN	40051	1	none	Temporary indices
iR	40052	1	none	Temporary indices
iWS	40053	1	none	Temporary indices
iWSA	40054	1	none	Temporary indices
iWSB	40055	1	none	Temporary indices
iWSC	40056	1	none	Temporary indices
j	40057	1	none	Temporary indices
jN	40058	1	none	Temporary indices
jP	40059	1	none	Temporary indices
jR	40060	1	none	Temporary indices
K	40061	1	none	Temporary indices
KDT	40062	1	none	Flag in CDT for $\Delta t$ change
KN	40063	1	none	Temporary indices
KP	40064	1	none	Temporary indices
KR	40065	1	none	Temporary indices
LRM	40066	1	none	Temporary indices
L	40067	1	none	Temporary indices
M	40068	1	none	Temporary indices
MA	40069	1	none	Temporary indices
MB	40070	1	none	Temporary indices
MC	40071	1	none	Temporary indices
MD	40072	1	none	Temporary indices
ME	40073	1	none	Temporary indices
MZ	40074	1	none	Set = 150 in input required in EDIT
N	40075	1	none	Temporary indices

<u>Symbol</u>	<u>Location</u>	<u>No. of Words</u>	<u>Units</u>	<u>Description</u>
REZ	40076	1	none	Not used
TRAD	40077	1	none	Not used
DTRAD	40078	1	none	Not used
RADEB	40079	1	none	Total positive Z momentum
RADER	40080	1	none	Total positive X momentum
RADET	40081	1	none	Total positive Y momentum
X1	40082	1	none	Not used
X2	40083	1	none	Not used
Y1	40084	1	none	Not used
Y2	40085	1	none	Not used
iMAXA	40086	1	none	Not used

### The Z Block

<u>Location</u>	<u>Symbol</u>	<u>Units</u>	<u>Description</u>
Z(1)	PROB	none	Problem number
Z(2)	CYCLE	none	Floating point value of the cycle number
Z(3)	DT	shake	$\Delta t_{hydro} = t^n - t^{n-1}$
Z(4)	PRINTS	none	Cycle frequency for short print
Z(5)	PRINTL	none	Cycle frequency for long print
Z(6)	DUMPT7	none	Cycle frequency for binary tape dump
Z(7)	CSTOP	none	Cycle number at which problem stops
Z(8)	PIDY	none	$= \pi = 3.1415927$
Z(9)	GAM	none	Not used
Z(10)	GAMD	none	$= 1. / (\alpha - 1.)$ computer in input
Z(11)	GAMX	none	$= 1. / (\alpha_x - 1.)$ routine
Z(12)	ETH	jerks	$=$ total energy in the system (originally set = to (iMAX)(jMAX)(kMAX))

$$\sum_{L=1}^{} AMX(L) [I(L) + \frac{1}{2} U^2(L) + V^2(L) + W^2(L)]$$

Changed to PH1 at transmittive boundaries and in PH2 if mass leaves the system

<u>Location</u>	<u>Symbol</u>	<u>Units</u>	<u>Description</u>
Z(13)	FFA	none	Upper limit for stability and used to calculate $\Delta t$ only if CABLN = 0.
Z(14)	FFB	none	Lower limit for stability and used to calculate $\Delta t$ only if CABLN = 0.
Z(15)	TMASS	none	Epsilonics on minimum specific internal energy
Z(16)	XMAX	none	Epsilonics on minimum velocity components
Z(17)	YMAX	none	Not used
Z(18)	ZMAX	none	Not used
Z(19)	DNN	none	$= (\text{ETH}-\text{E}^{\text{N}-\text{NPC}}/\text{ETH})$ for energy check
Z(20)	DMIN	none	if ECK (see definition in Z(24)) is > DMIN, problem will stop because of energy violation
Z(21)	DTNA	shake	$\Delta t^{n-1}$
Z(22)	REZFCT	none	Not used
Z(23)	TOZONE	g/cm <sup>3</sup>	If the mass flow across a free surface within the grid produces a density < TOZONE, the mass flow is set to zero
Z(24)	ECK	none	Is the energy check $(\text{ETH}-\text{E}^{\text{n}}/\text{ETH}) - (\text{ETH}-\text{E}^{\text{N}-\text{NPC}}/\text{ETH}) / \text{NPC}$ Where NPC = cycle frequency at which the energy check is made
Z(25)	SBOUND	none	Fractions of $\Delta$ in mass weighted velocity (suggested number = 1.0)
Z(26)	CABLN	none	Defined in Section 5.1
Z(27)	T	shake	Total time up to cycle NC $t^{n+1} = t^n + \Delta t$
Z(28)	GMAX	none	Maximum of the two gammas ( $\gamma_x$ or $\gamma$ )
Z(29)	WSGD	none	$\gamma$ .
Z(30)	WSGX	none	$\gamma_x$ and ( $\gamma_{\text{MAX}-1.}$ ) in CDT routine
Z(31)	GMADR	none	$\gamma/(\gamma-1.)$
Z(32)	GMAXP	none	$\gamma_x/(\gamma_x-1.)$

<u>Location</u>	<u>Symbol</u>	<u>Units</u>	<u>Description</u>
Z(33)	$\rho_o$	$g/cm^3$	
Z(34)	a	none	
Z(35)	$E_o$	jerks/g	
Z(36)	b	none	
Z(37)	A	jerks/cm <sup>3</sup>	
Z(38)	$V_s$	none	
Z(39)	none	none	
Z(40)	$E_s$	jerks/g	
Z(41)	$\alpha$	none	
Z(42)	$\beta$	none	
Z(43)	0	none	
Z(44)	B	jerks/cm <sup>3</sup>	
Z(45)	DTCHK	$g/cm^3$	Density checks, if $\rho(L) < DTCHK$ , the stability check for cell (L) will be bypassed
Z(46)	PCSTAB	none	% of stability, used if CABLN is < 0. A recommended value is .25
Z(47)	CNOT	$10^5 \text{ cm/sec}$	
Z(48)	BFACT	none	
Z(49)	EPSI	none	
Z(50)	S1	none	
Z(51)	S2	none	
Z(52)	S3	none	
Z(53)	S4	none	
Z(54)	S5	none	
Z(55)	S6	none	
Z(56)	S7	none	
Z(57)	S8	none	Used in set-up routine
Z(58)	S9	none	Not used
Z(59)	S10	none	Initial X velocity component of the projectile in cm/shake
Z(60)	AMLOST	g	Similar term for the Z direction
Z(61)	ELOST	jerks	Mass thrown away (PH2)
Z(62)	XMLOST	g cm/shake	Energy thrown away (PH2)
Z(63)	YMLOST	g cm/shake	Total X momenta thrown away (PH2)
Z(64)	ZMLOST	g cm/shake	Total Y momenta thrown away (PH2)
Z(65)	ENEG	jerks	Total Z momenta thrown away (PH2)
Z(66)	RHONOT	$g/cm^3$	Energy added to system if I < 0.
Z(67)	VELOC	cm/shake	Initial density of material
Z(68)	BUG	none	Initial velocity of pellet in the Y direction (cm/shake)
			Epsilon for emptying pellet (~ .01)

<u>Location</u>	<u>Symbol</u>	<u>Units</u>	<u>Description</u>
Z(69)		none	Defined in Section 5.1
Z(70)		none	Defined in Section 5.1
Z(71)		none	j value of top of target
Z(72)		none	Not used
Z(73)		none	Not used
Z(74)		none	Not used
Z(75)		none	Defined in Section 5.1
Z(76)		none	Defined in Section 5.1
Z(77)		none	Defined in Section 5.1
Z(78)		none	Defined in Section 5.1
Z(79)		none	Defined in Section 5.1
Z(80)		none	Defined in Section 5.1
Z(81)	NPR	none	Index (working storage)
Z(82)	NPRI	none	Index (working storage)
Z(83)	NC	none	Cycle number in fixed point
Z(84)	NPC	none	Number or cycles between energy checks
Z(85)	NRC	none	Index
Z(86)	iMAX	none	Maximum number of zones in X direction
Z(87)	jMAX	none	Maximum number of zones in Y direction
Z(88)	kMAX	none	Maximum number of zones in Z direction
Z(89)	kMAXA	none	Total number of zones = (iMAX)(jMAX)(kMAX) = (iMAX)(jMAX)
Z(90)	iXMAX	none	Index (working storage)
Z(91)	NOD	none	Index (working storage)
Z(92)	NOPR	none	Index (working storage)
Z(93)	i1	none	Minimum value of i in do loop on i
Z(94)	i2	none	Maximum value of i in do loop on i
Z(95)	i3	none	Original interface between projectile and target

<u>Location</u>	<u>Symbol</u>	<u>Units</u>	<u>Description</u>
Z(96)	i4	none	Number of zones to the right for rezone
Z(97)	N1	none	Flag at left, if 0., boundary is transmittive, otherwise reflective
Z(98)	N2	none	Flag at right, if 0., boundary is transmittive, otherwise reflective
Z(99)	N3	none	Flag at top, if 0., boundary is transmittive, otherwise reflective
Z(100)	N4	none	Flag at bottom, if 0., boundary is transmittive, otherwise reflective
Z(101)	N5	none	Flag behind, if 0., boundary is transmittive, otherwise reflective
Z(102)	N6	none	Flag in front, if 0., boundary is transmittive, otherwise reflective
Z(103)	N7	none	Binary tape number designation
Z(104)	N8	none	Not used
Z(105)	N9	none	k value of zone that is controlling $\Delta t$
Z(106)	N10	none	i value of zone that is controlling $\Delta t$
Z(107)	N11	none	j value of zone that is controlling $\Delta t$
Z(108)	k1	none	Minimum value of k in do loop on k
Z(109)	k2	none	Maximum value of k in do loop on k
Z(110)	j1	none	Minimum value of j in do loop on j
Z(111)	j2	none	Maximum value of j in do loop on j
Z(112) through Z(150)			Not used

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2. Dienes, J. K., W. E. Johnson and J. M. Walsh, "Annual Status Report on the Theory of Hypervelocity Impact," GA-6509. June 28, 1965.
3. Harlow, F. H., "Two-Dimensional Hydrodynamic Calculations," LA-2301. September 1959.
4. Johnson, W. E., "TOIL (A Two-Material Version of the OIL Code)," GAMD-8073. July 13, 1967.